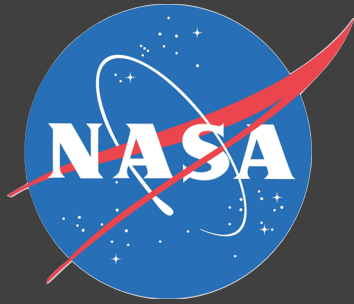


# CART3D SIMULATIONS

FOR THE 2ND AIAA SONIC BOOM PREDICTION WORKSHOP



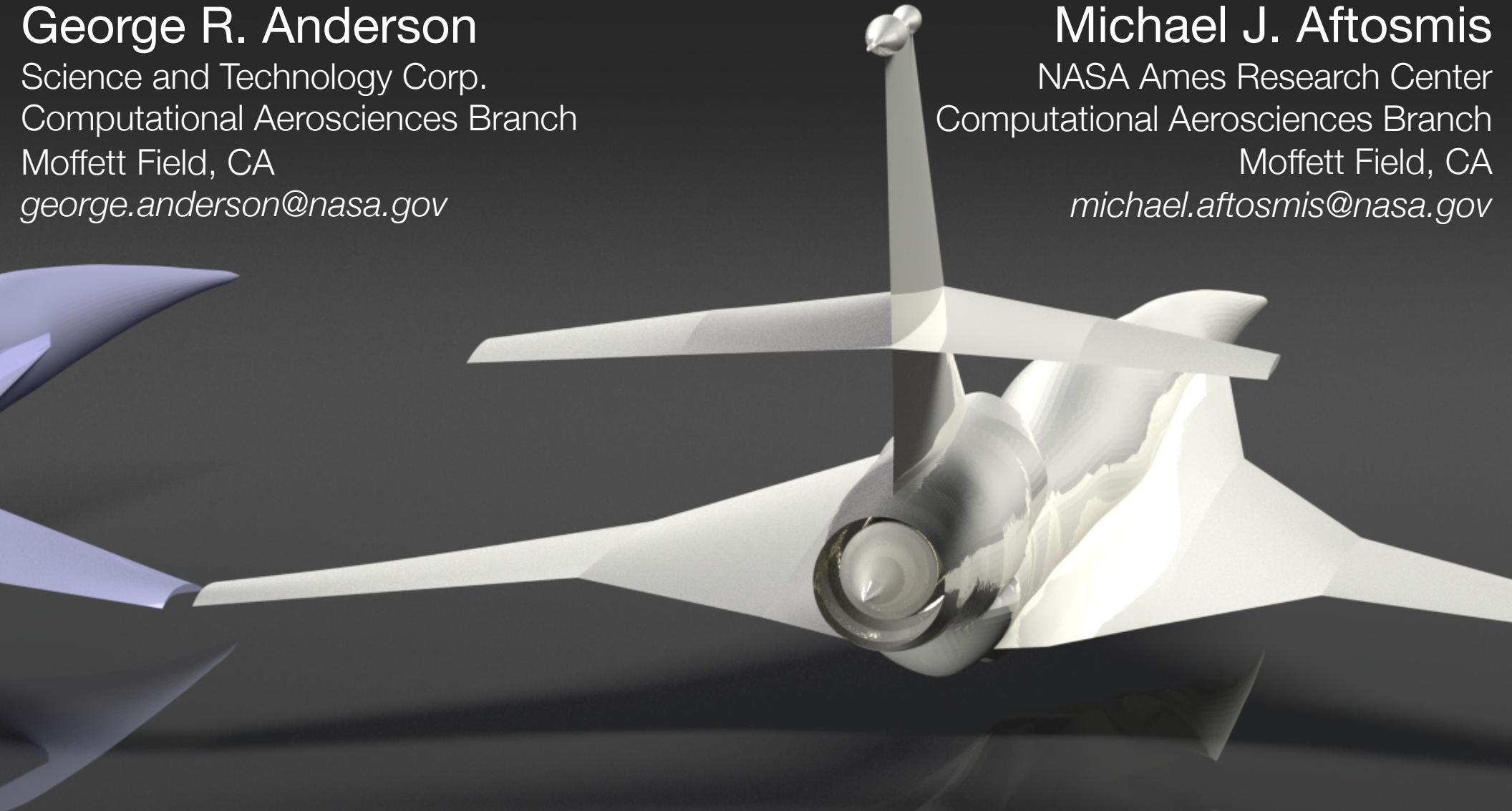
7 JANUARY 2016, GRAPEVINE, TEXAS

**George R. Anderson**

Science and Technology Corp.  
Computational Aerosciences Branch  
Moffett Field, CA  
*george.anderson@nasa.gov*

**Michael J. Aftosmis**

NASA Ames Research Center  
Computational Aerosciences Branch  
Moffett Field, CA  
*michael.aftosmis@nasa.gov*





# CASES SUBMITTED

**AXIBODY**

**JWB**

**C25-D**

*flow-through*

*powered*

ALL CASES:

- ▶  $M_\infty = 1.6$
- ▶ Altitude 15.76 km

**SUBMITTED:**

- ▶ All 4 cases, all azimuths, 3 mesh refinement levels
- ▶ Propagated signals and loudness metrics

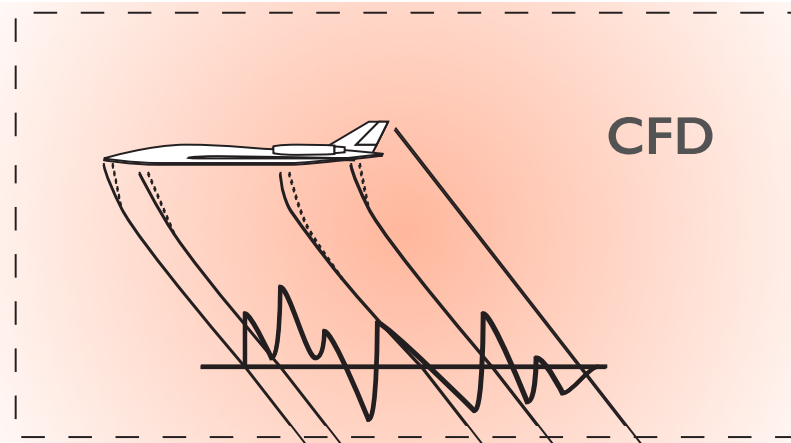
- ▶ **Solvers and Adaptive Meshing**  
*Axibody results as demo*
- ▶ **Multi-azimuth Problems with Cartesian Meshes**
- ▶ **Selected Workshop Results**
- ▶ **Mesh Size and Cost**
- ▶ **Conclusions**



# APPROACH

## Flow Solver — Cart3D v1.5

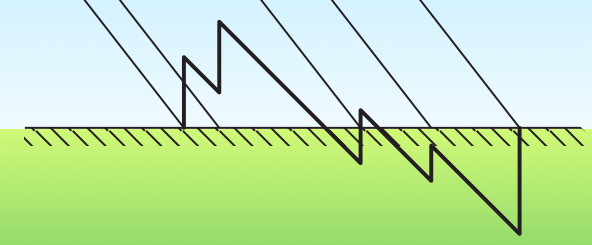
- ▶ Steady, inviscid flow
- ▶ Adjoint-based mesh refinement



## Propagation — sBOOM\* v2.5

- ▶ Augmented Burgers' Equation
- ▶ Standard atmosphere model

Atmospheric Propagation



## Loudness — LCASB\*\* v. 2009-02-27

- ▶ “Loudness Code for Asymmetric Sonic Booms”
- ▶ Computes Perceived Loudness (PLdB)

\* Rallabhandi, “Advanced Sonic Boom Prediction Using the Augmented Burgers Equation” J. Aircraft, 2011.

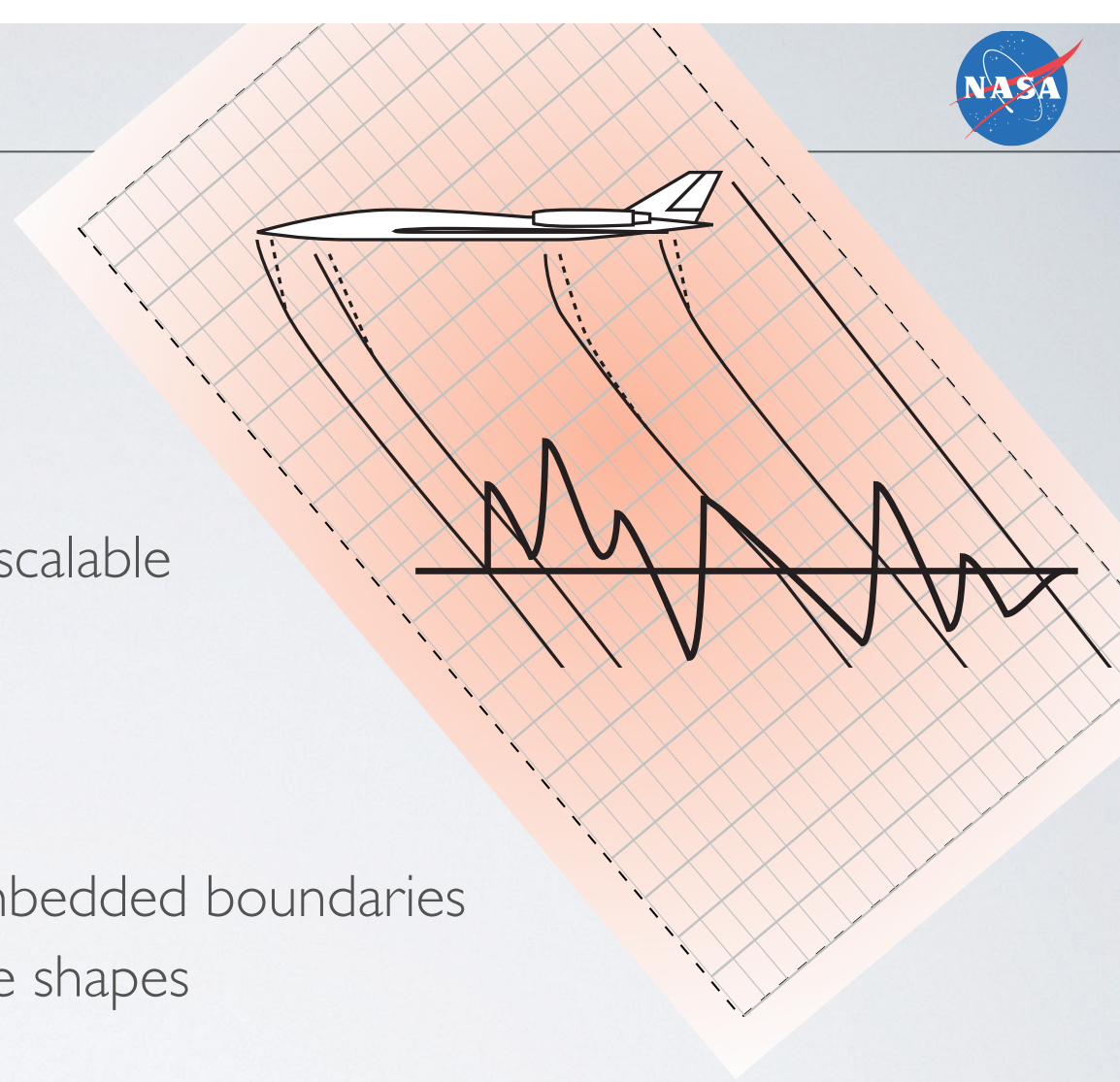
\*\* Shepherd and Sullivan, “A Loudness Calculation Procedure Applied to Shaped Sonic Booms”, NASA TP, 1991.



# CFD AND MESHING

## Flow Solver — Cart3D v1.5

- ▶ Steady, inviscid flow
- ▶ 2nd-order upwind method
- ▶ Multigrid acceleration
- ▶ Domain decomposition — highly scalable
- ▶ *For today: van Leer, Barth-Jespersen*



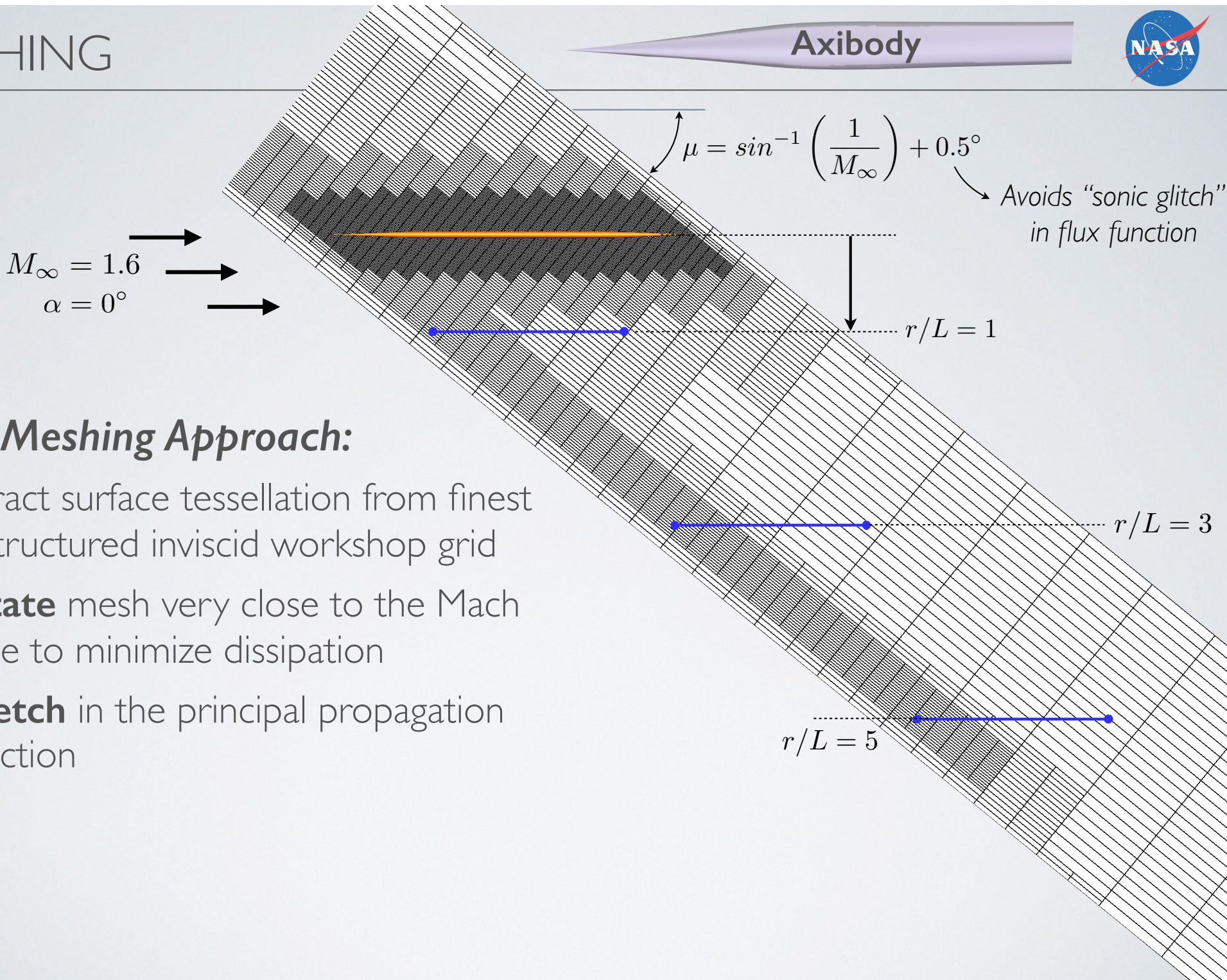
## Automatic Meshing

- ▶ Multilevel Cartesian mesh with embedded boundaries
- ▶ Handles arbitrarily complex vehicle shapes

## Output-Driven Mesh Adaptation

- ▶ Local error estimates computed via discrete adjoint
- ▶ Mesh is refined in locations with highest impact on pressure signatures

# MESHING



## Basic Meshing Approach:

- ▶ Extract surface tessellation from finest unstructured inviscid workshop grid
- ▶ **Rotate** mesh very close to the Mach angle to minimize dissipation
- ▶ **Stretch** in the principal propagation direction





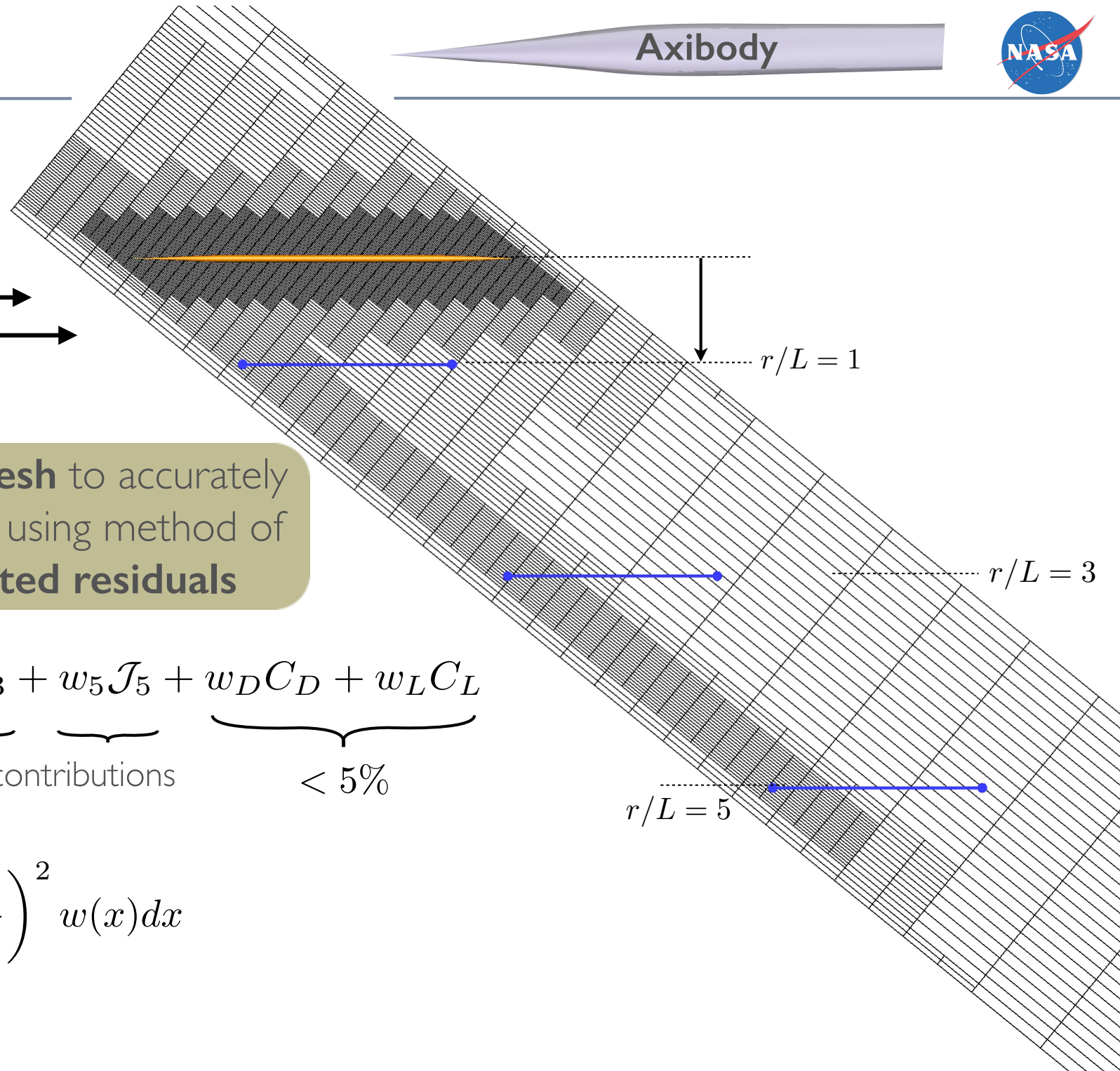
$M_\infty = 1.6$   
 $\alpha = 0^\circ$

**Locally refine mesh** to accurately compute outputs, using method of **adjoint-weighted residuals**

$$\mathcal{J} = w_1 \mathcal{J}_1 + w_3 \mathcal{J}_3 + w_5 \mathcal{J}_5 + w_D C_D + w_L C_L$$

Roughly equal contributions  $< 5\%$

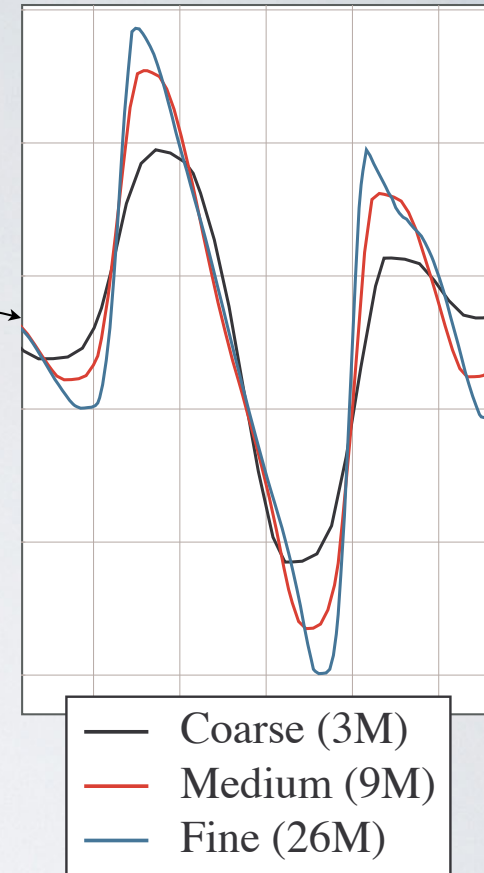
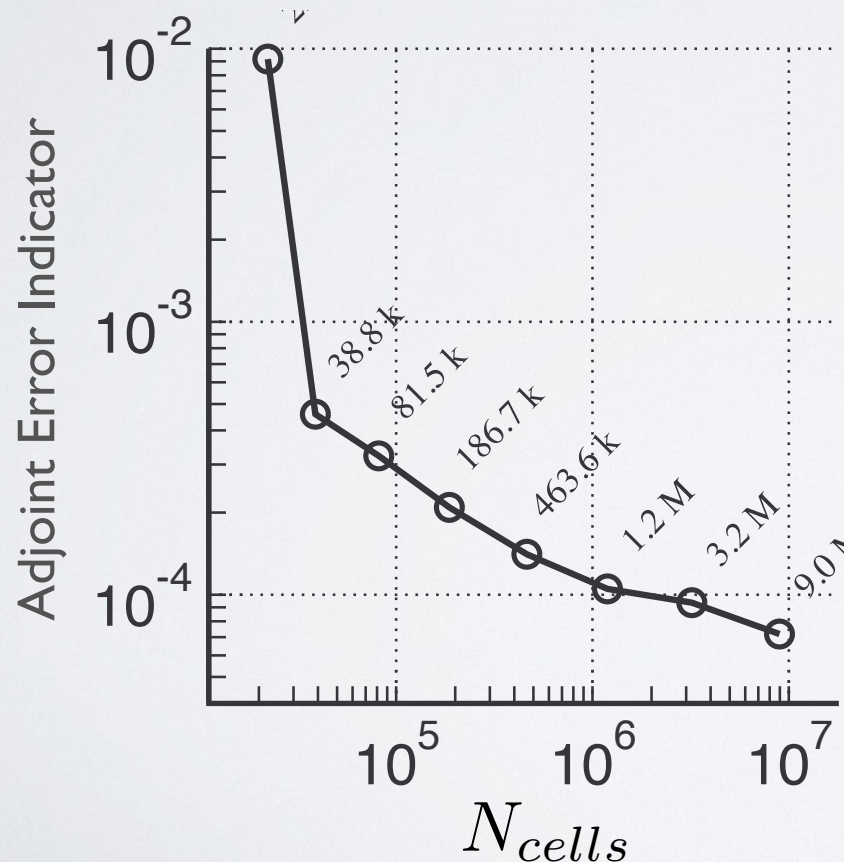
$$\mathcal{J}_i = \int_0^L \left( \frac{p - p_\infty}{p_\infty} \right)^2 w(x) dx$$



# MESH CONVERGENCE GUIDELINES

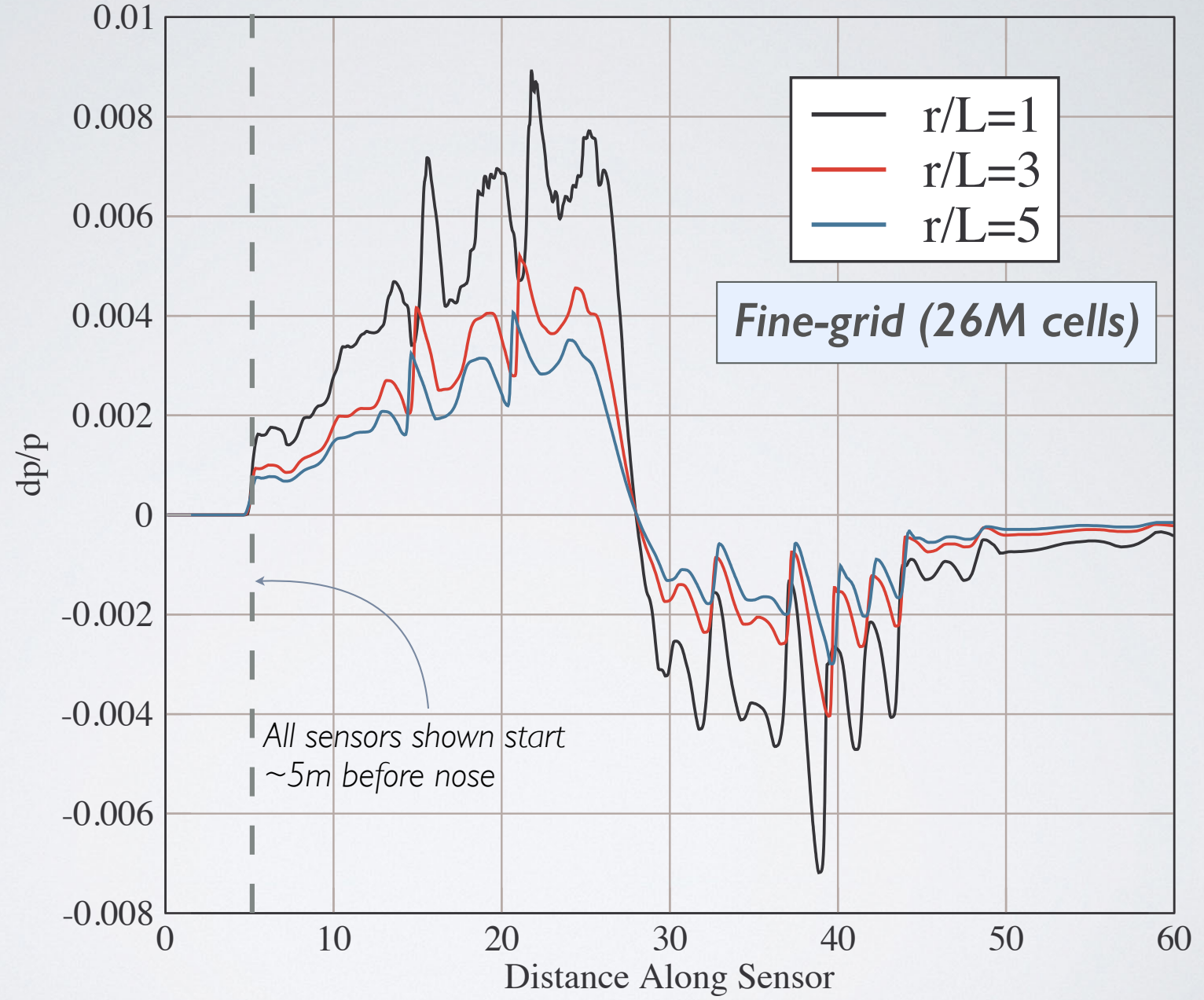
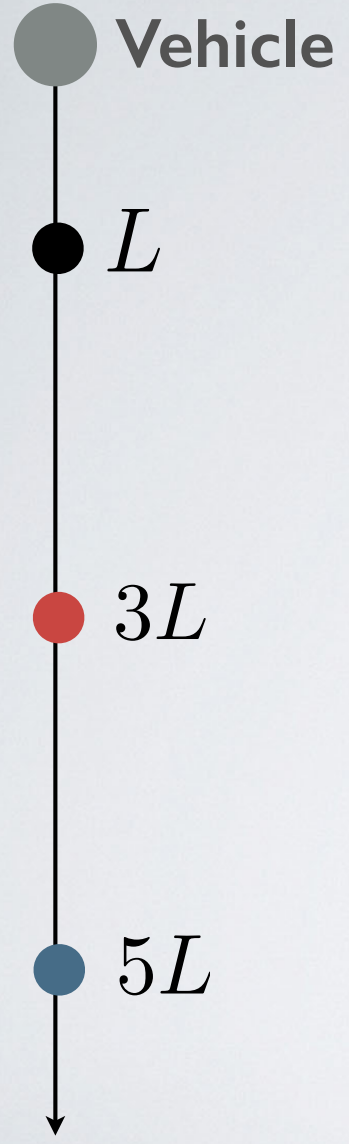
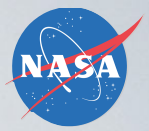
## What do we call “coarse”, “medium” and “fine”?

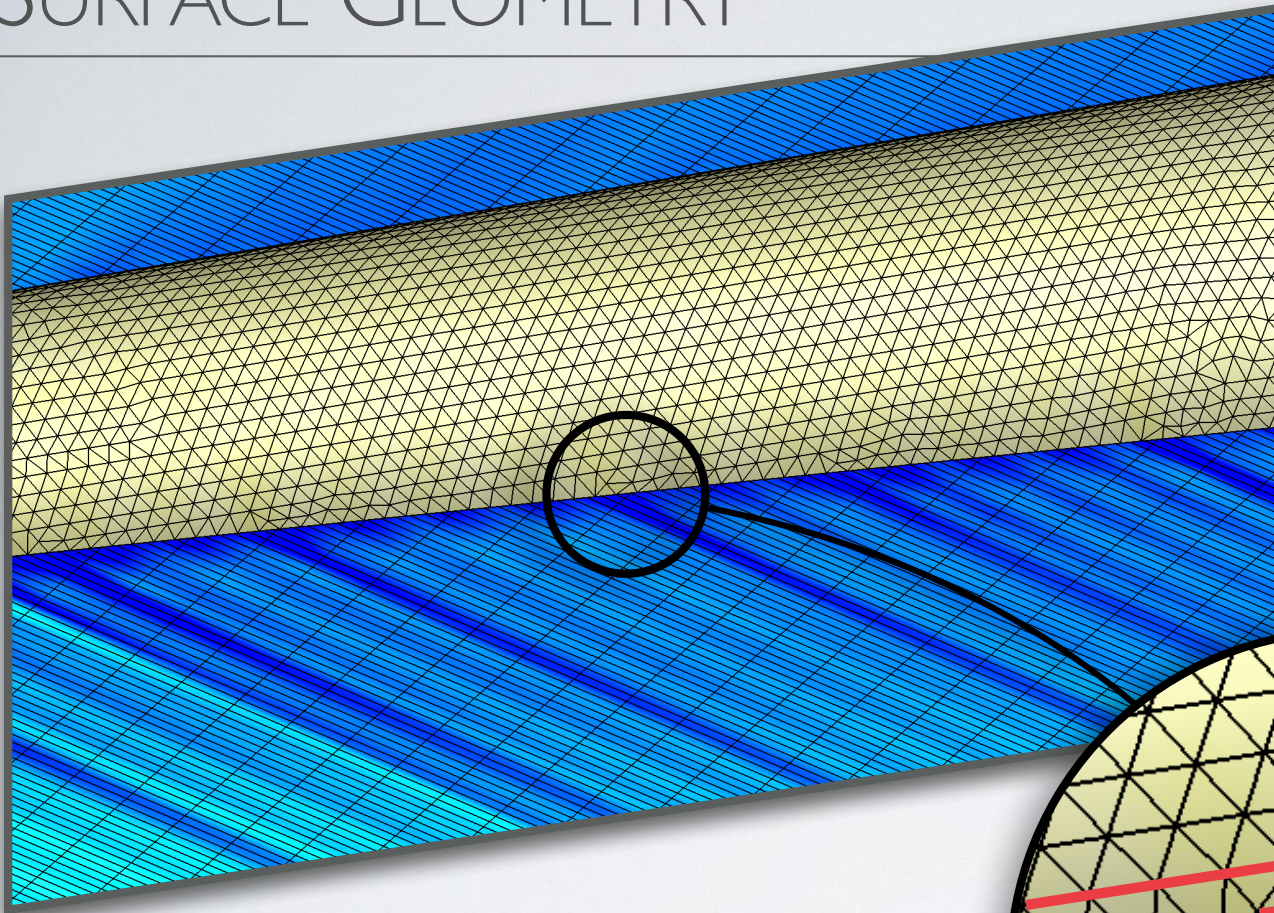
- ▶ Aware of workshop mesh sizes and guidelines, but allow adaptation to discover unexpected features!
- ▶ Guidelines we used:
  - ▶ **Qualitative:** Consistent signal features over consecutive meshes
  - ▶ **Quantitative:** Convergence of functional and **error** with refinement





# AXIBODY — SIGNALS



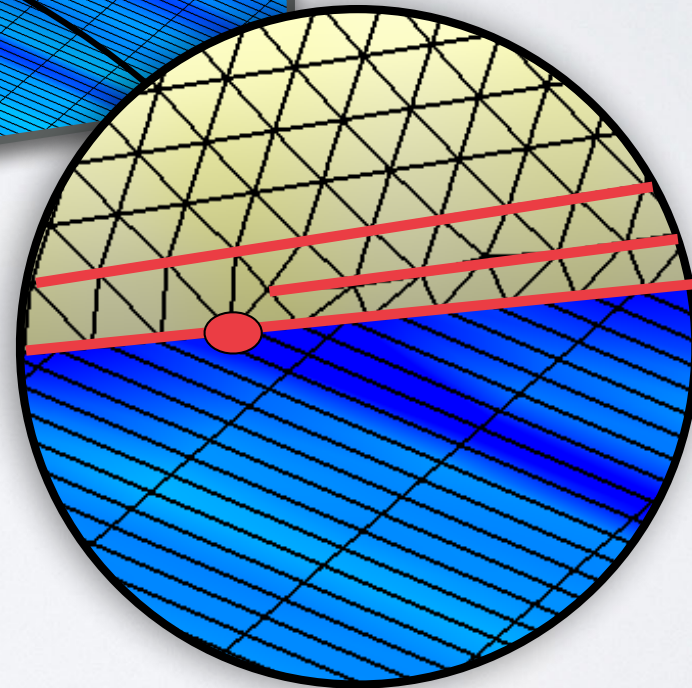


## Workshop Surface Mesh

(Boundary of workshop “fine”, inviscid, unstructured grid)

- ▶ Relatively coarse for Cart3D
- ▶ Irregular tessellation, not perfectly axisymmetric

**Irregular vertices**  
generate pressure  
disturbances

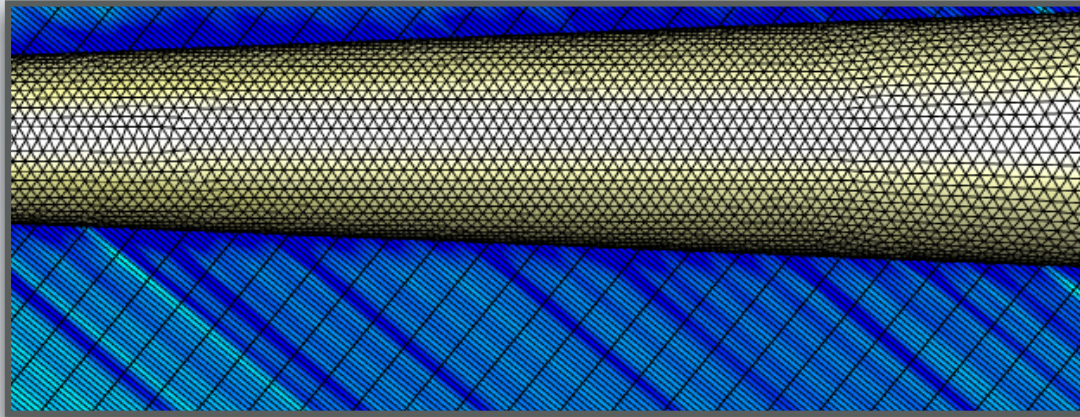




# SURFACE GEOMETRY

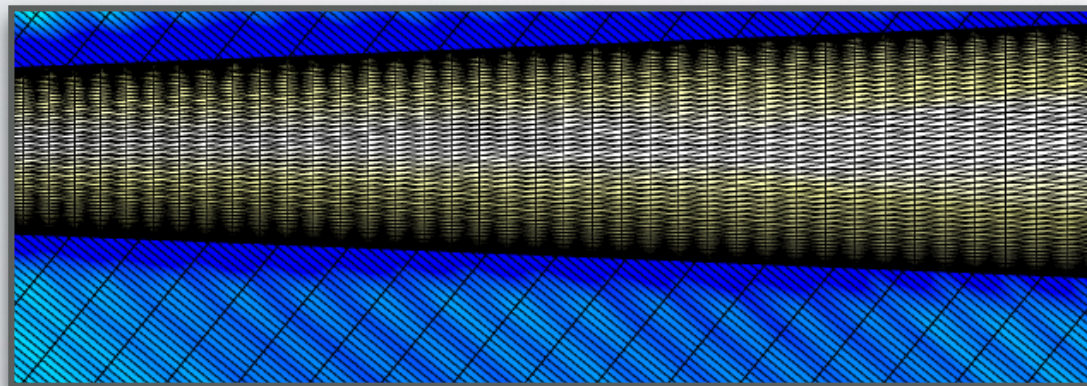
## Workshop Grid

- ▶ Irregular oscillations, **persist through  $r/L=1$ .**

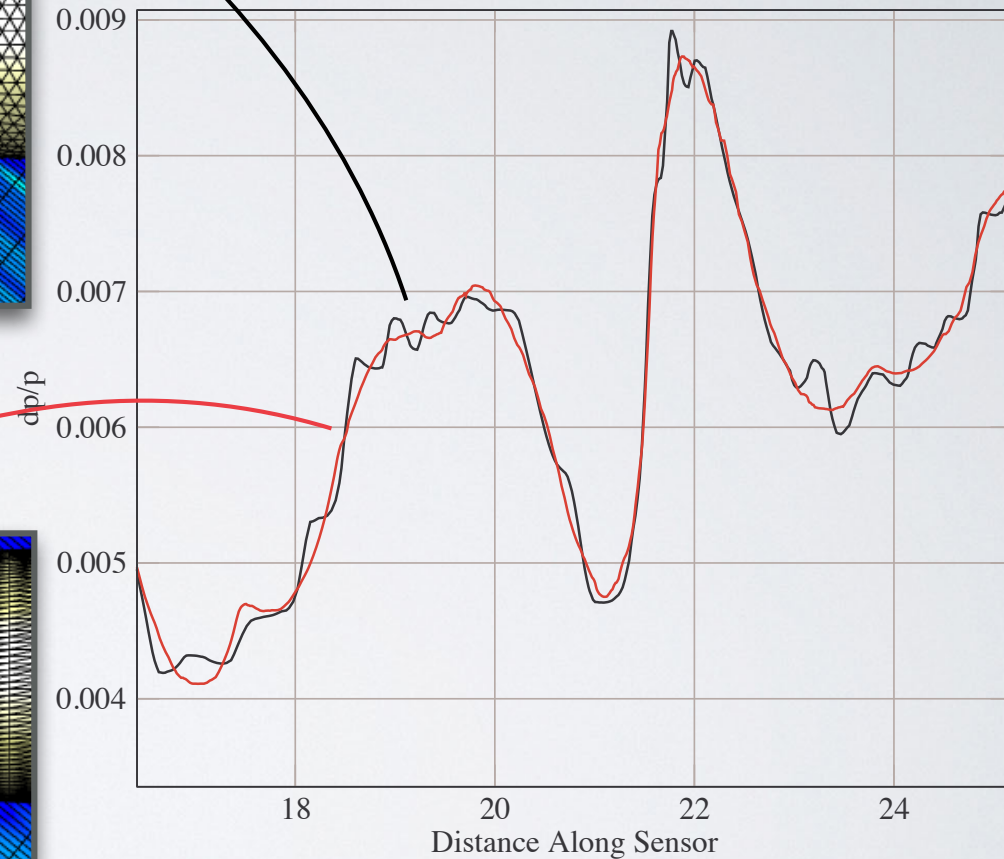


## In-house Grid

- ▶ Regular oscillations, **cancel quickly.**



Fine-grid Signatures at  $r/L=1$

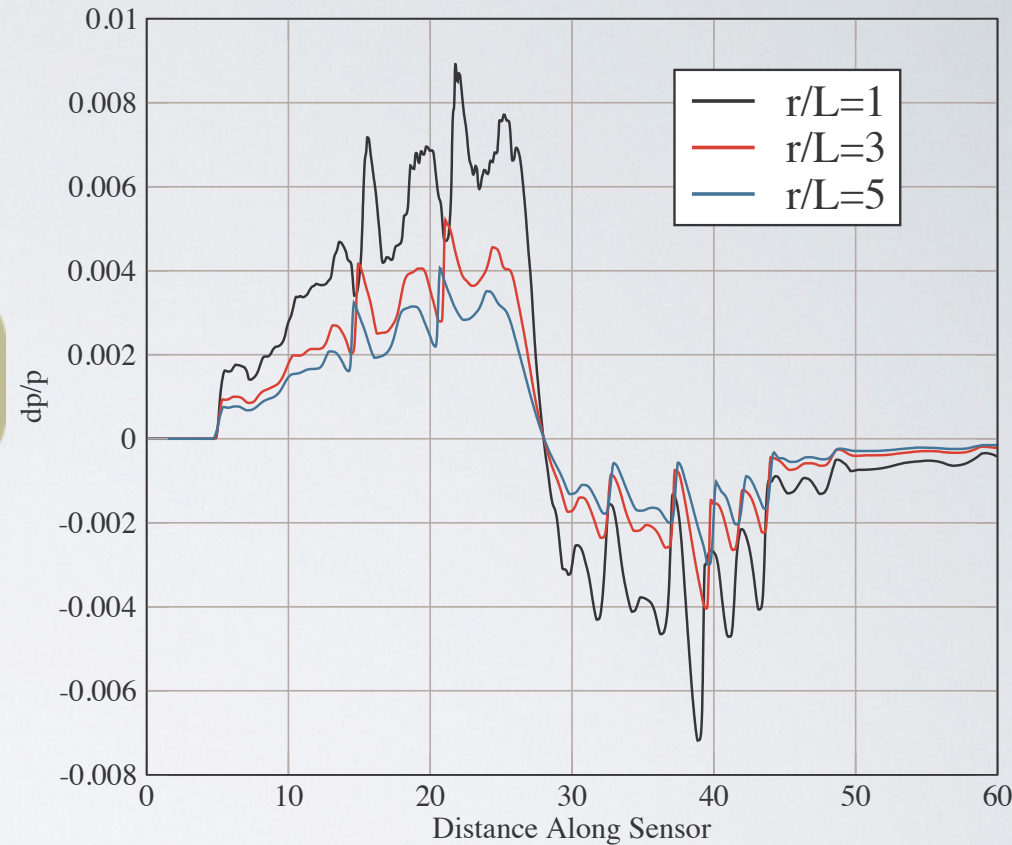


# SURFACE GEOMETRY

- ▶ At the farther distances, the oscillations disappear, even using the workshop meshes.

All results submitted use the provided workshop surface grids

- ▶ **Downside:** Near-body signals are somewhat oscillatory
- ▶ **Upside:** Consistency with workshop, faster than generating in-house grids.

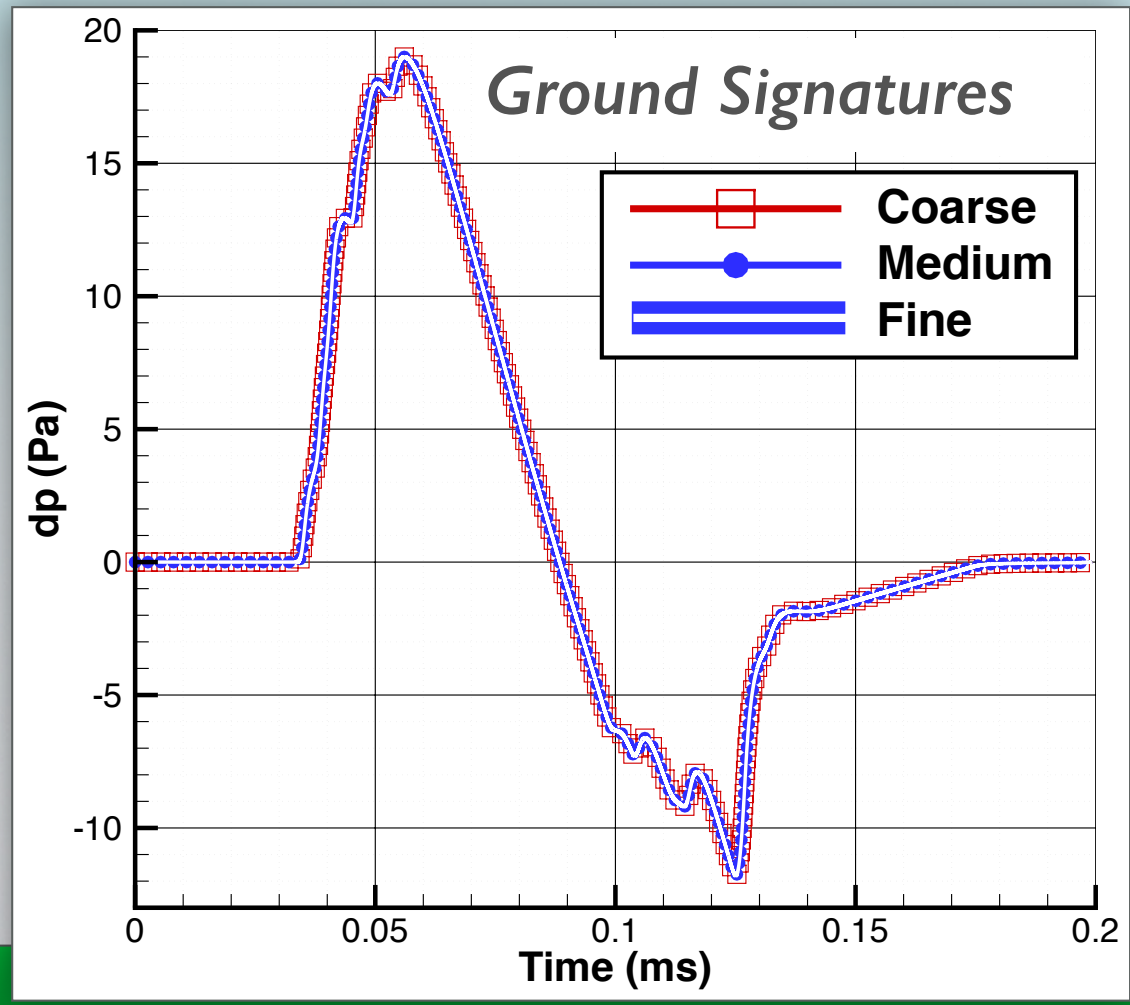
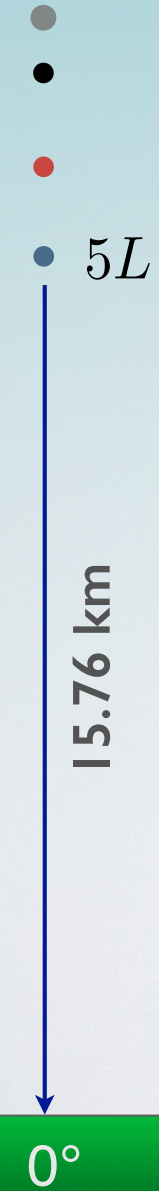




# AXIBODY — PROPAGATION AND LOUDNESS

## Propagation with sBOOM

Std. atmosphere model, no wind

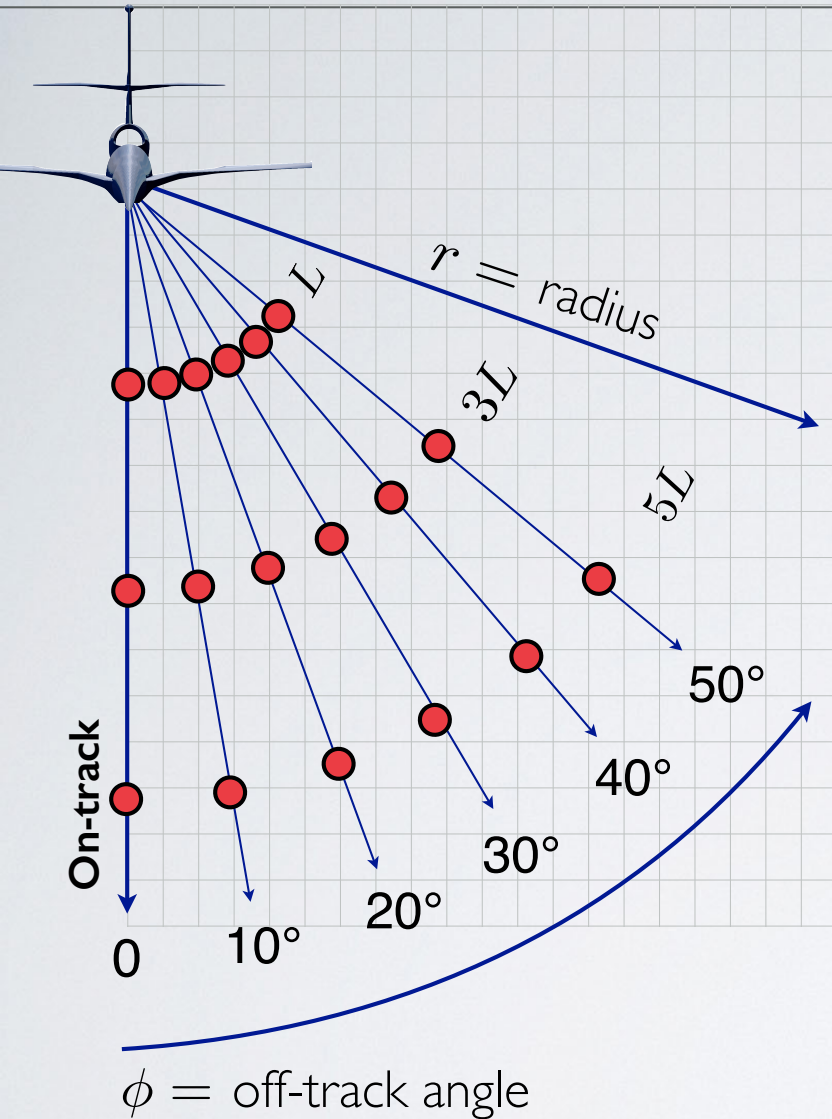


## Perceived Loudness (PLdB) (LCASB)

78.5	78.3	78.1
COARSE	MEDIUM	FINE
10MIN	30MIN	1HR   5MIN
CFD time		

Loudness is fairly insensitive to CFD resolution

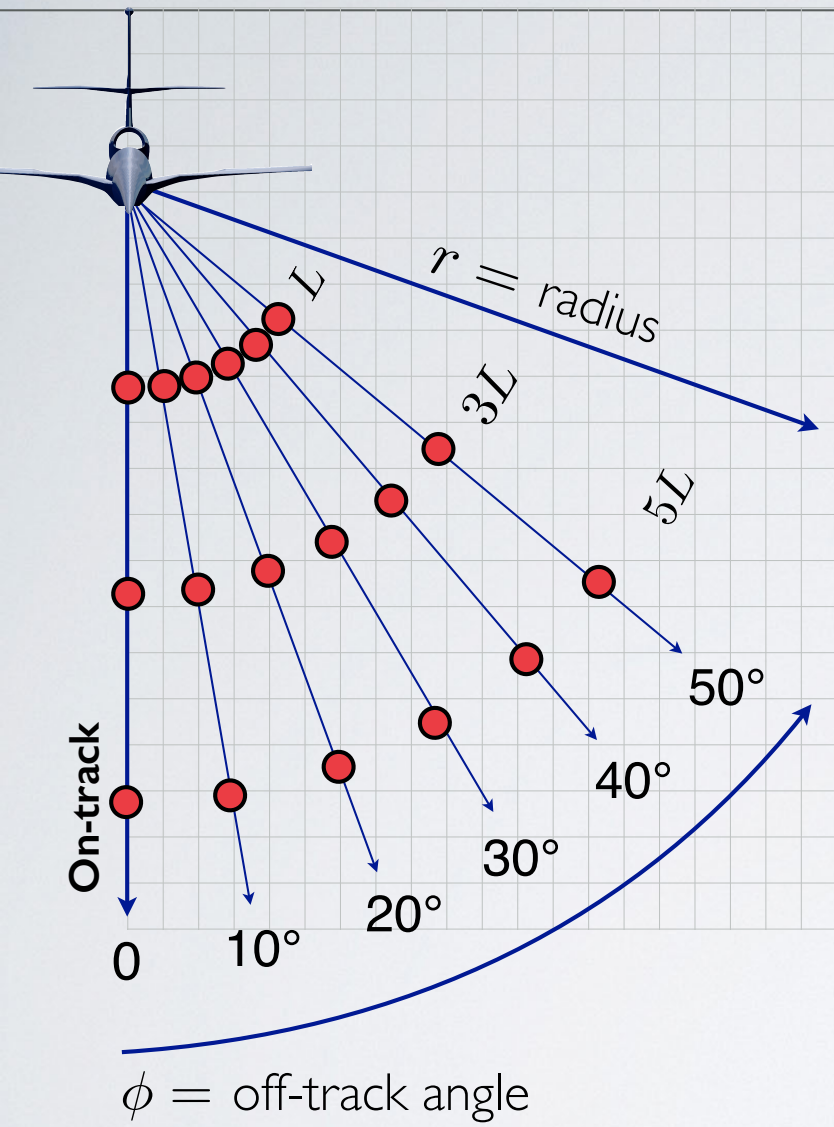
# OFF-TRACK SOLUTIONS



- ▶ Straightforward approach — **compute all sensors with a single mesh**
- ▶ With Cartesian-aligned grids, off-track angles are misaligned, leading to high dissipation and **high cell-counts**.

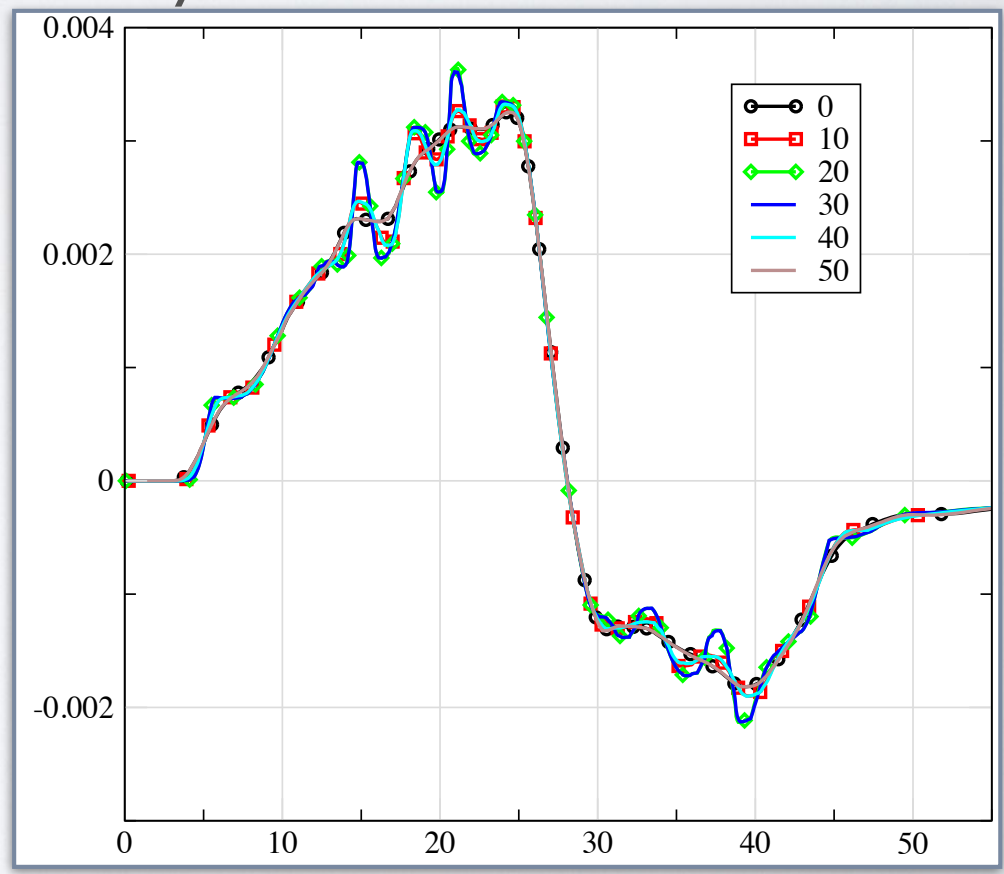


# OFF-TRACK SOLUTIONS



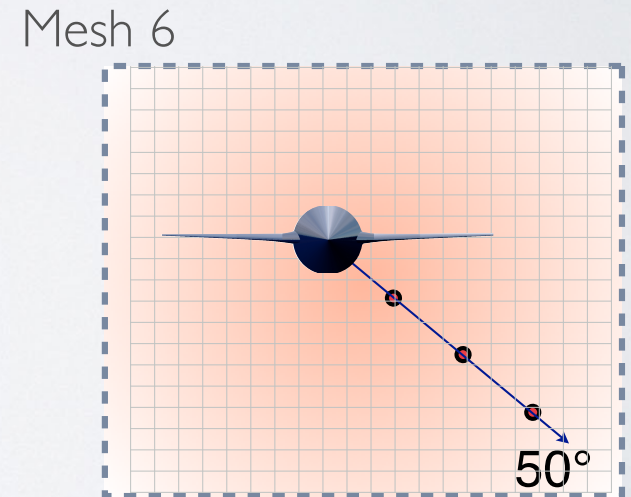
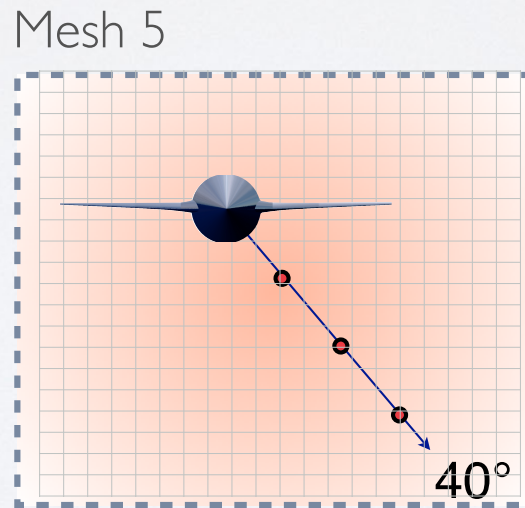
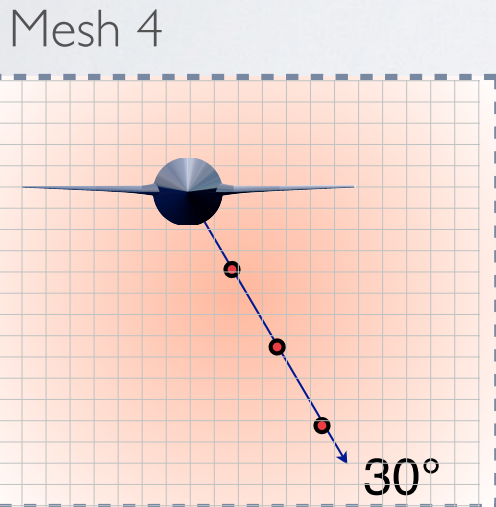
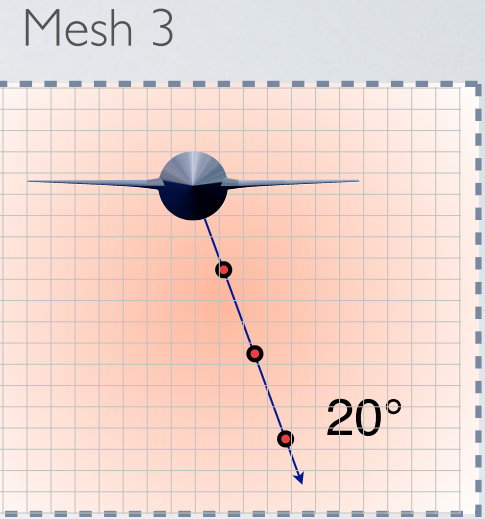
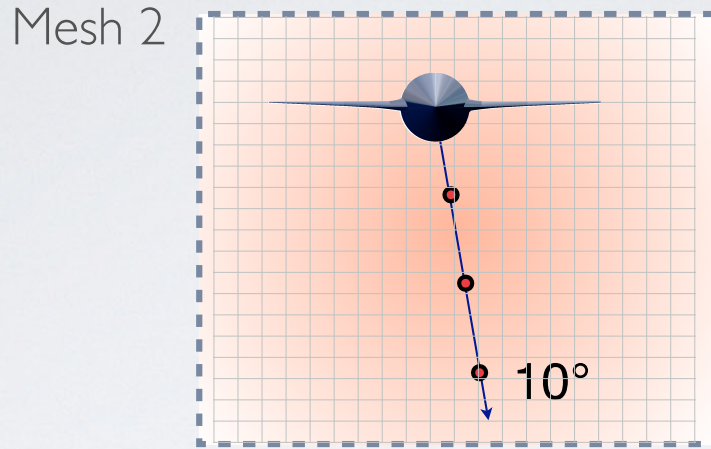
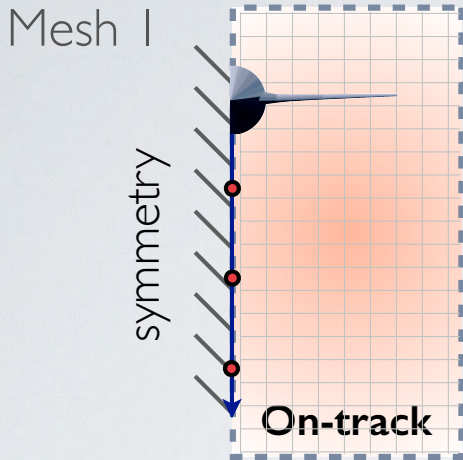
- ▶ Straightforward approach — **compute all sensors with a single mesh**
- ▶ With Cartesian-aligned grids, off-track angles are misaligned, leading to high dissipation and **high cell-counts**.

*Axibody case at all azimuths in one mesh*



# OFF-TRACK SOLUTIONS

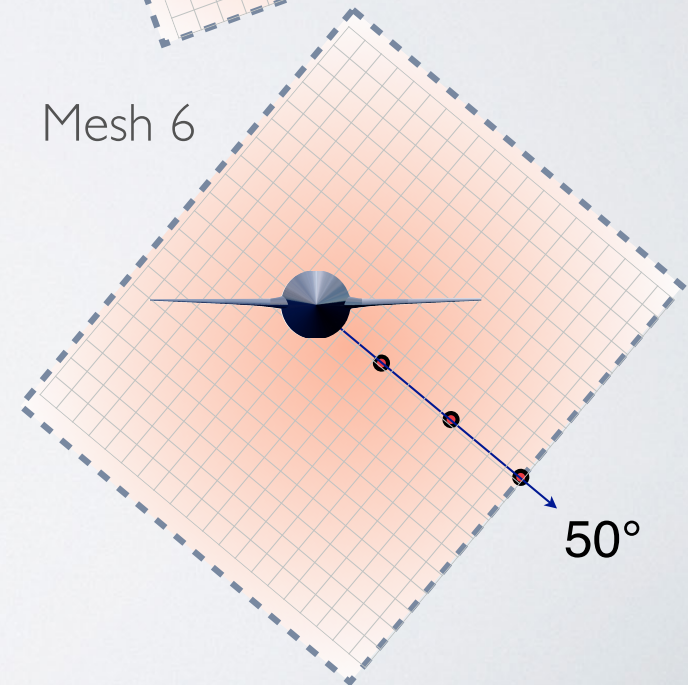
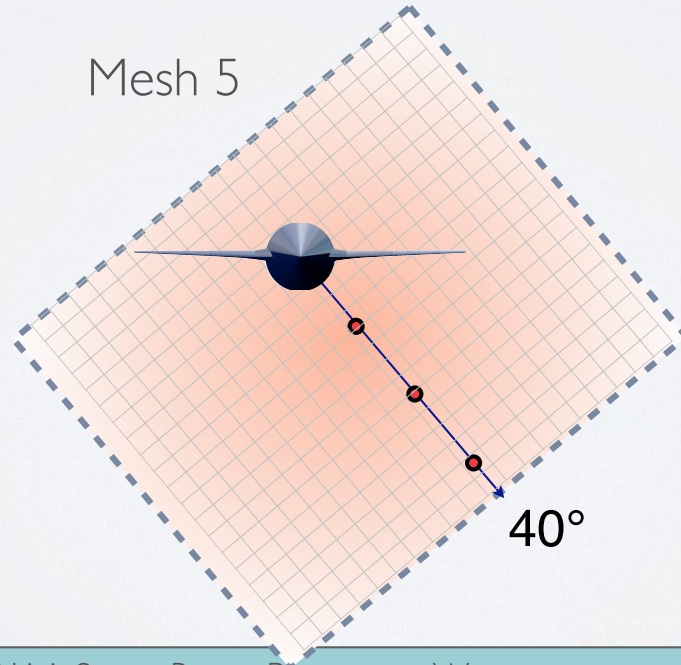
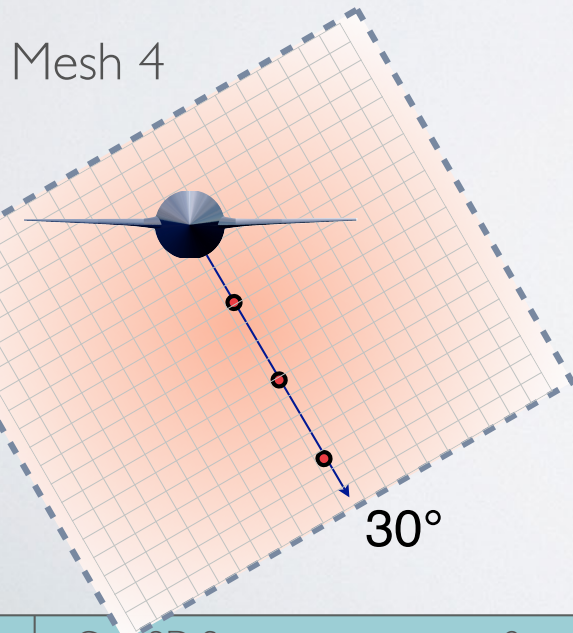
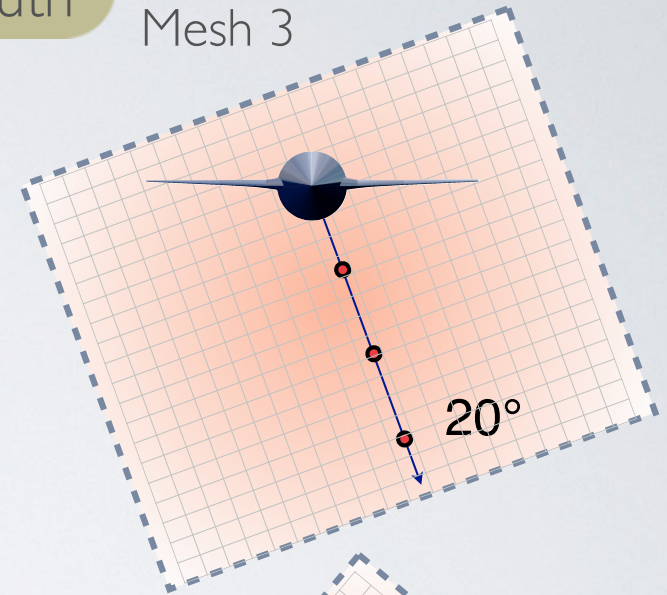
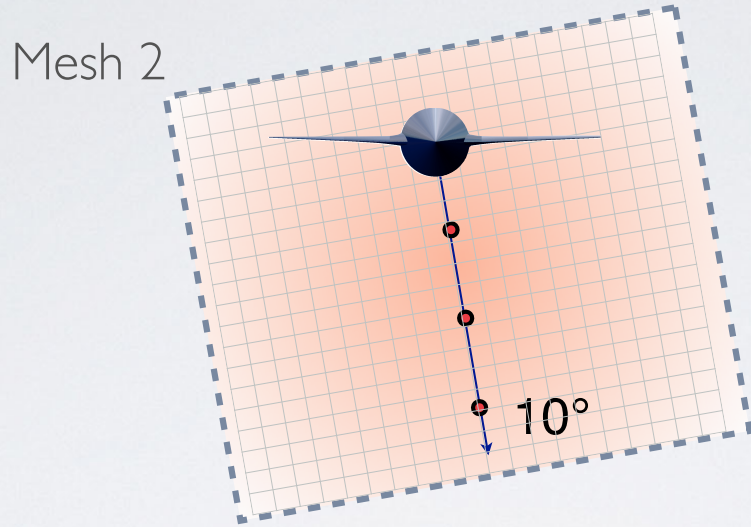
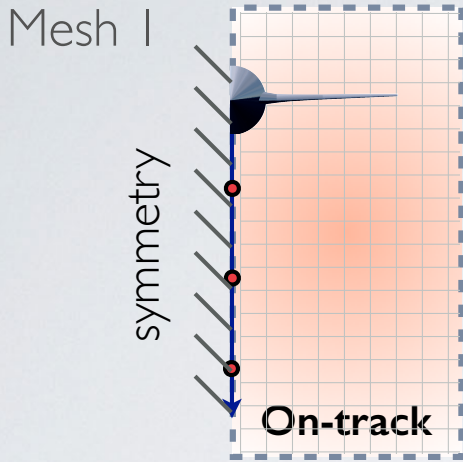
Use independent meshes,  
each rotated to off-track azimuth





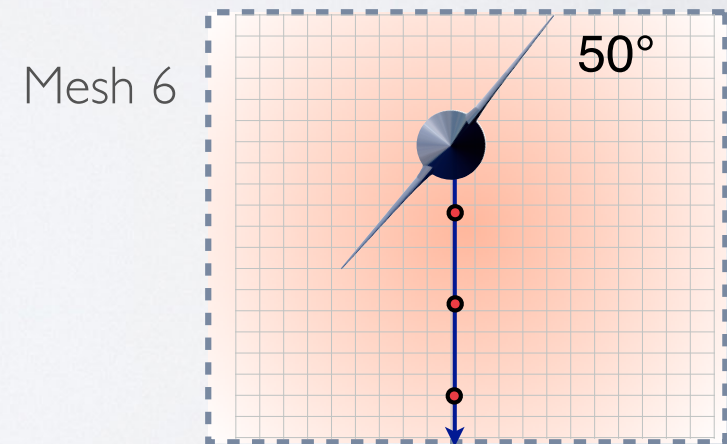
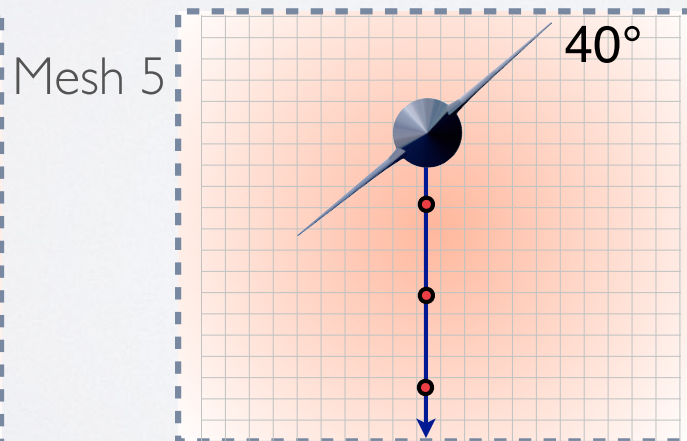
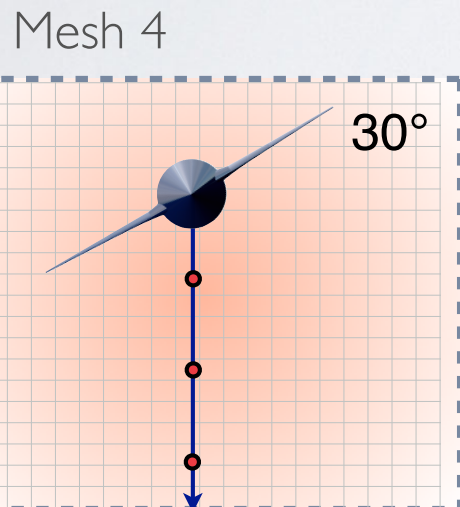
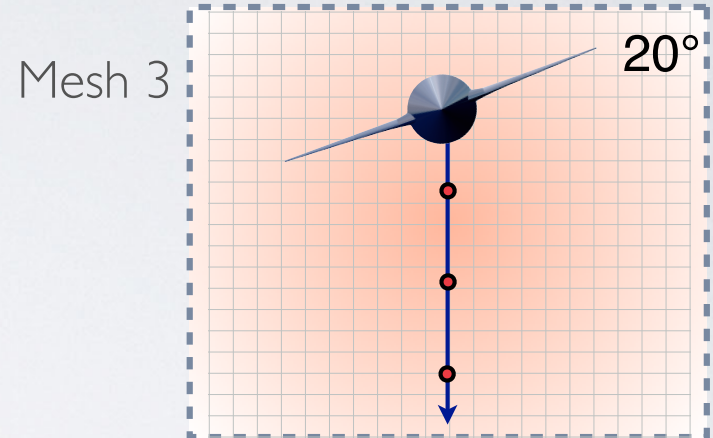
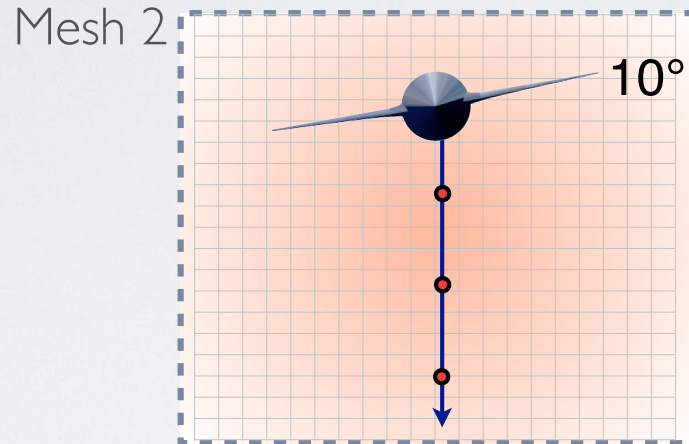
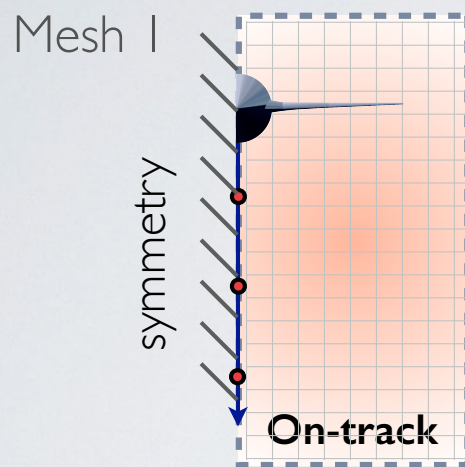
# OFF-TRACK SOLUTIONS

Use independent meshes,  
each rotated to off-track azimuth



# OFF-TRACK SOLUTIONS

- ▶ **Better quality** solutions — off-track signals are well-aligned with propagation
- ▶ **More efficient** — alignment permits much higher aspect ratio stretching
- ▶ **More parallel** — run each azimuth on a separate compute node







## ✓ **Solvers and Adaptive Meshing**

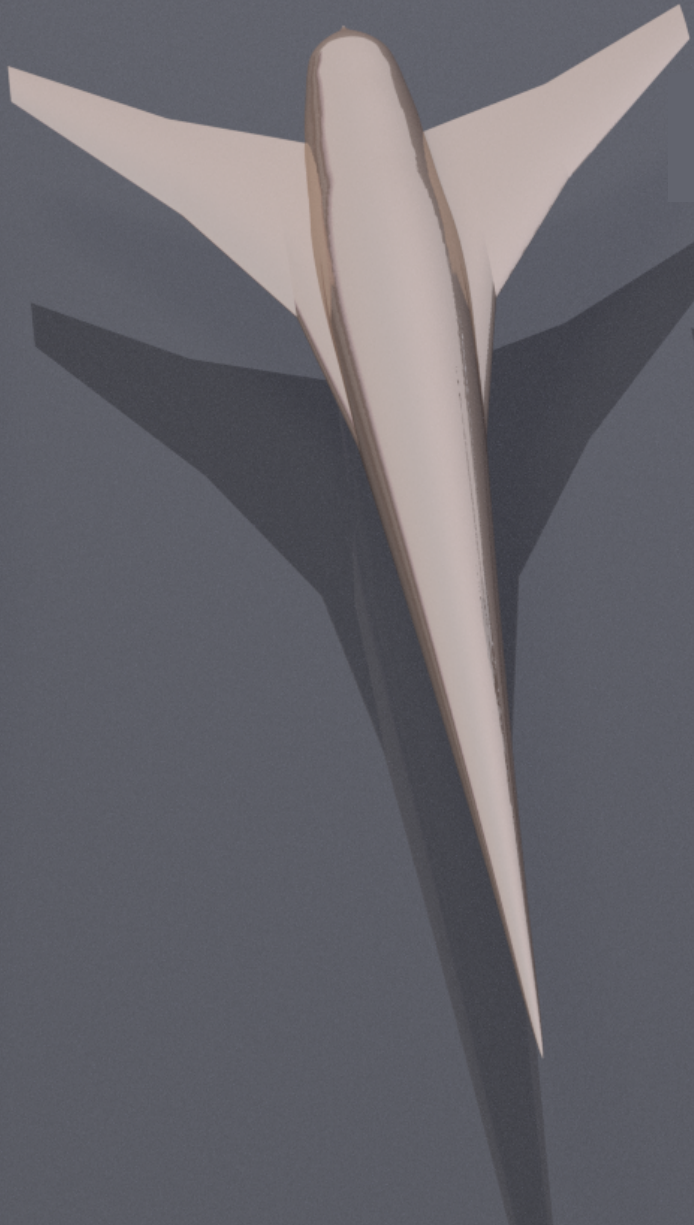
*Axibody results as demo*

## ✓ **Multi-azimuth Problems with Cartesian Meshes**

### ▶ **Selected Workshop Results**

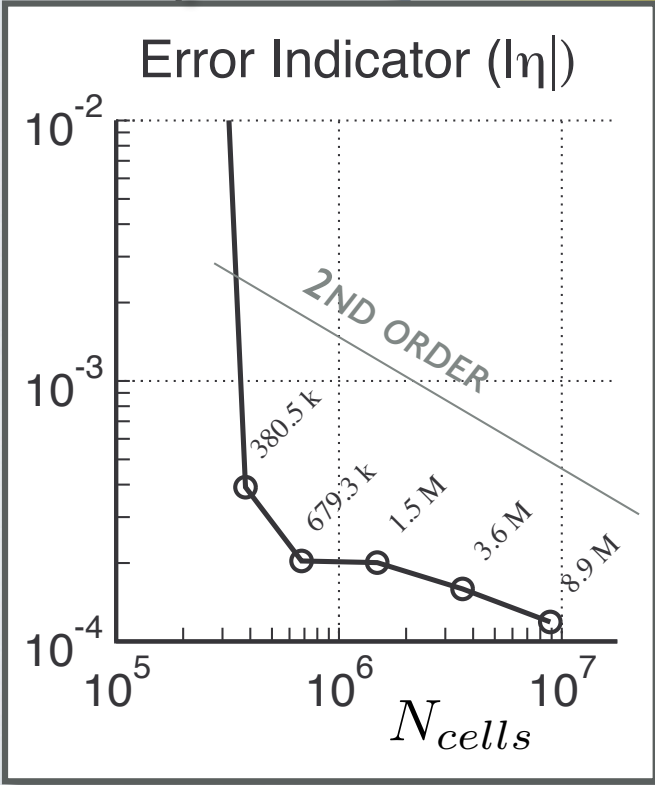
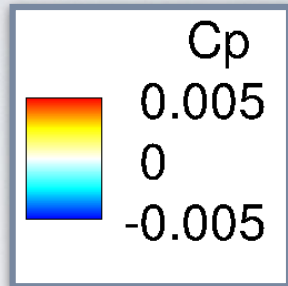
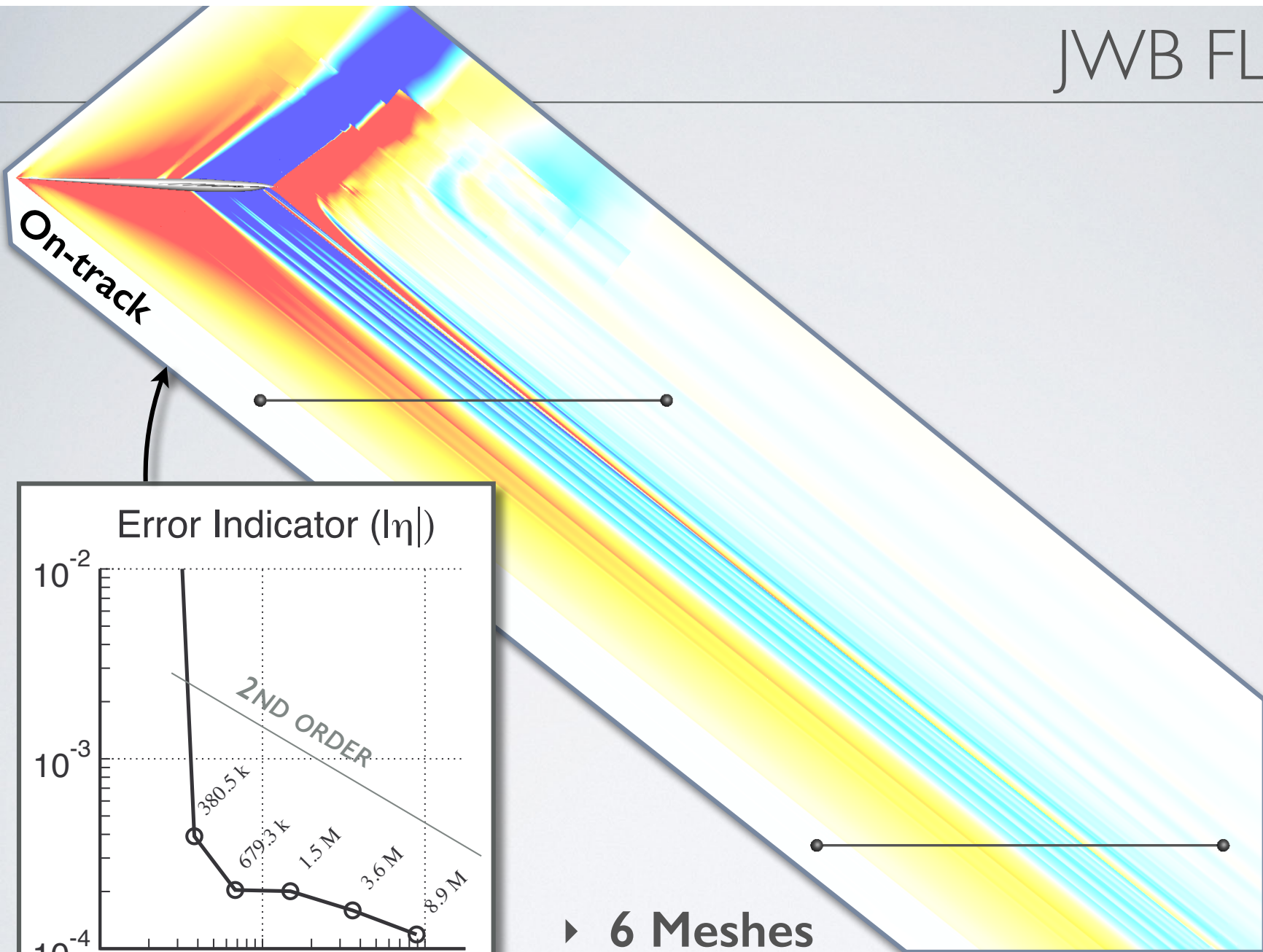
- Mesh Size and Cost
- Conclusions

# JAXA WING-BODY (JWB)

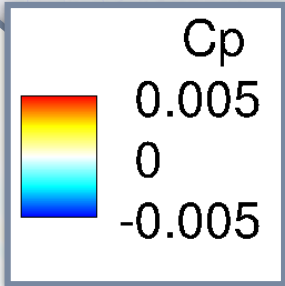
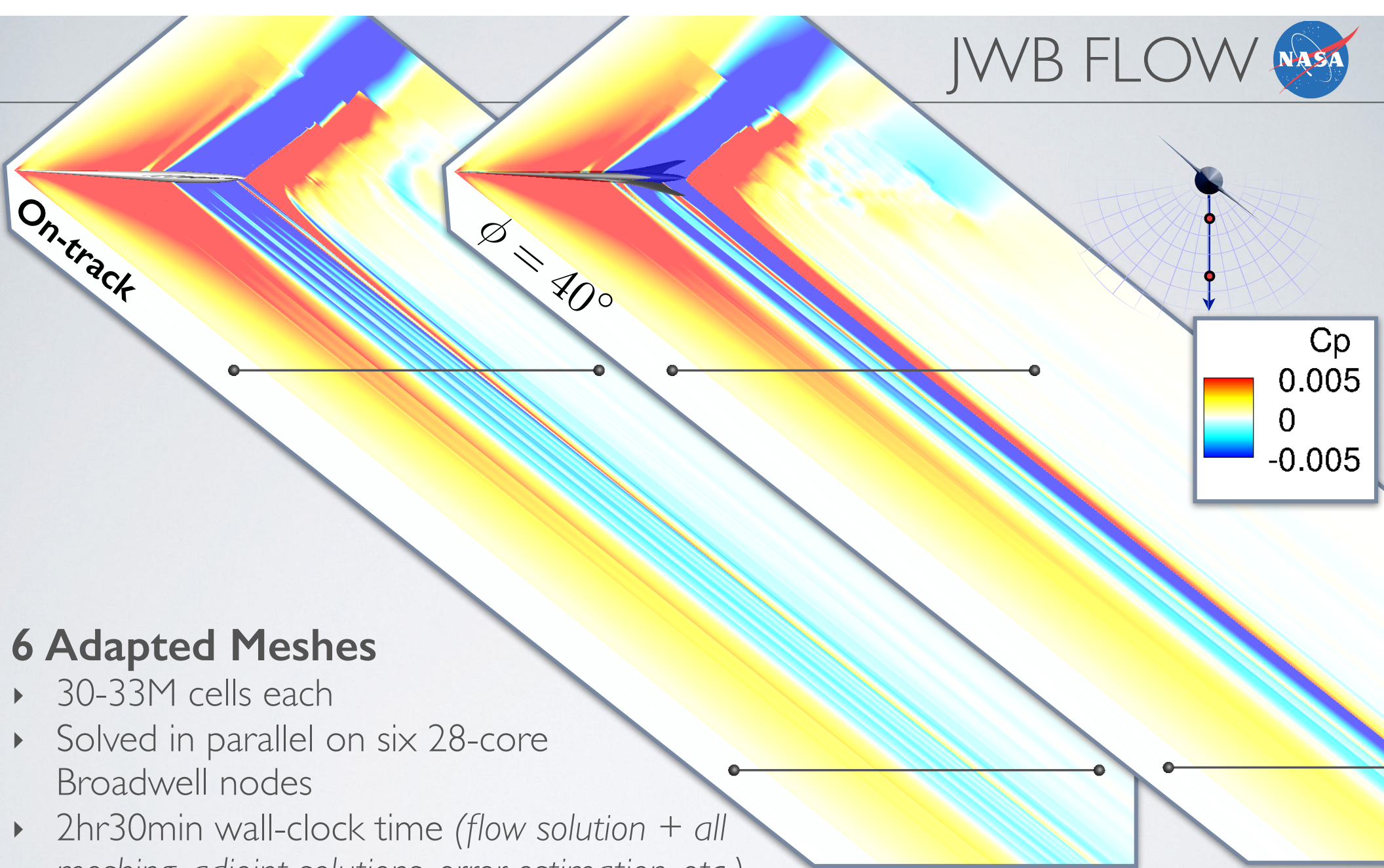


- ▶ Mach 1.6
- ▶  $\alpha = 2.3067^\circ$
- ▶ Computed  $C_L \approx 0.077$





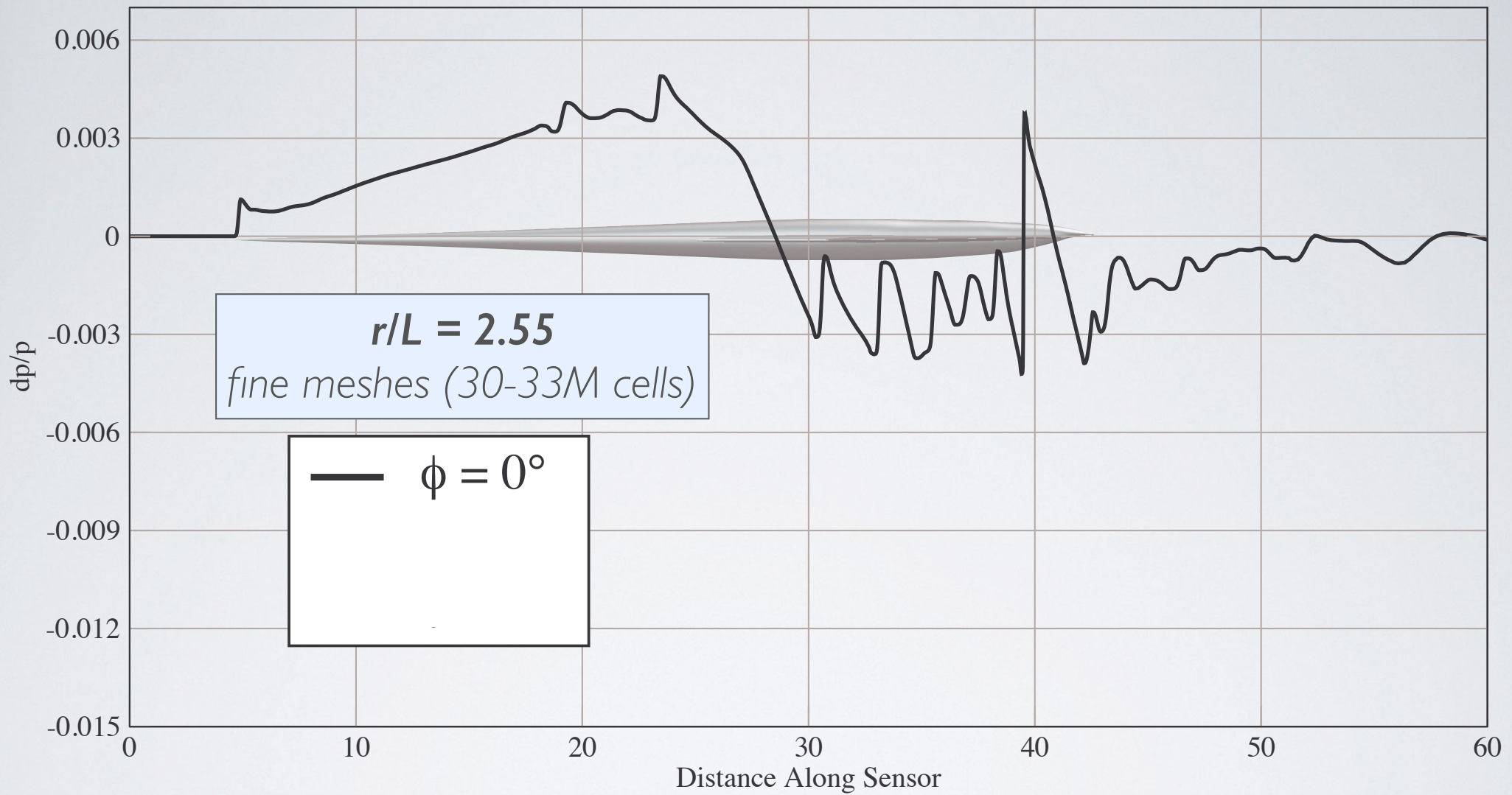
- ▶ **6 Meshes**
  - ▶ 30-33M cells each, independently adapted

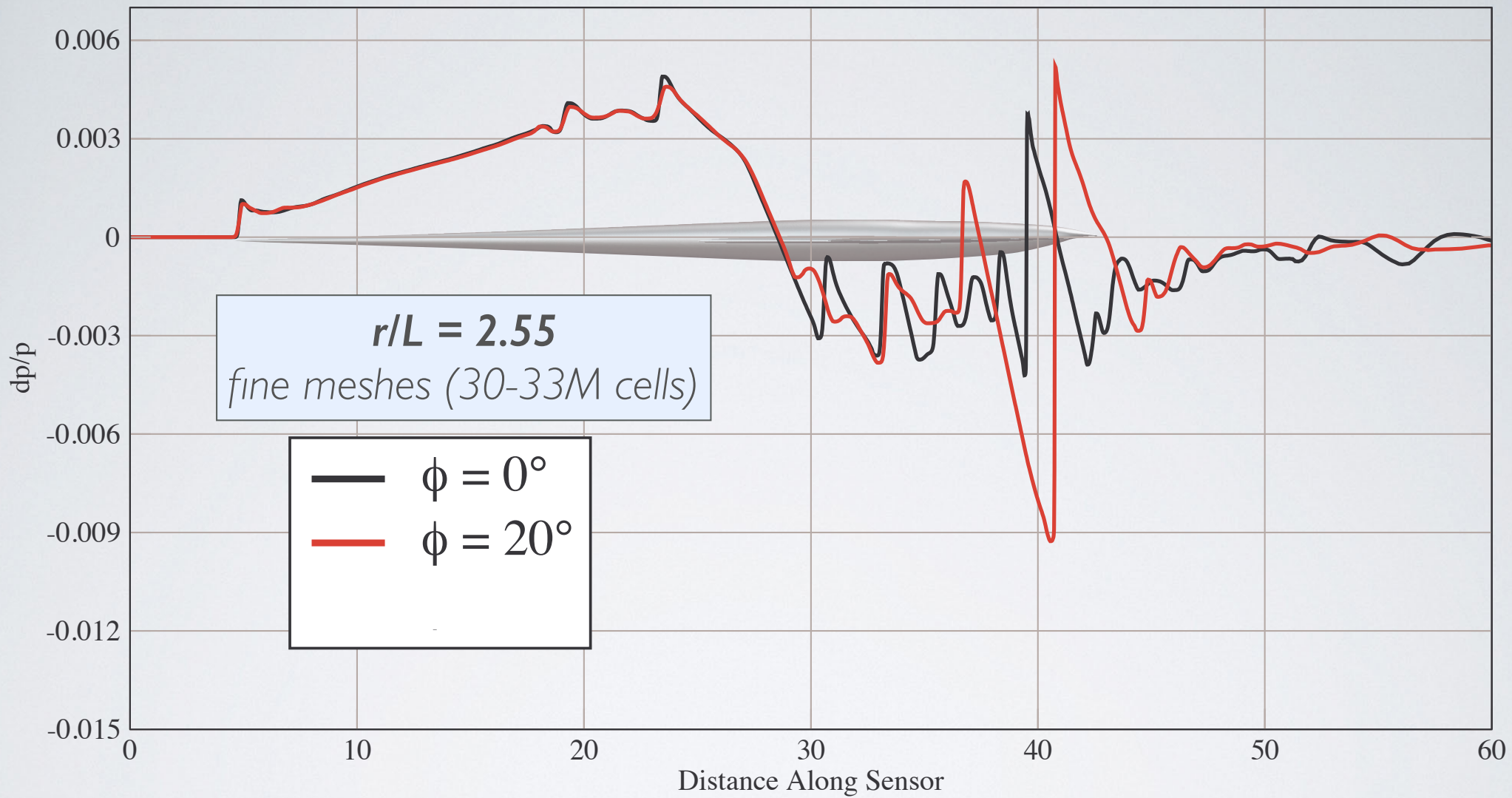


## 6 Adapted Meshes

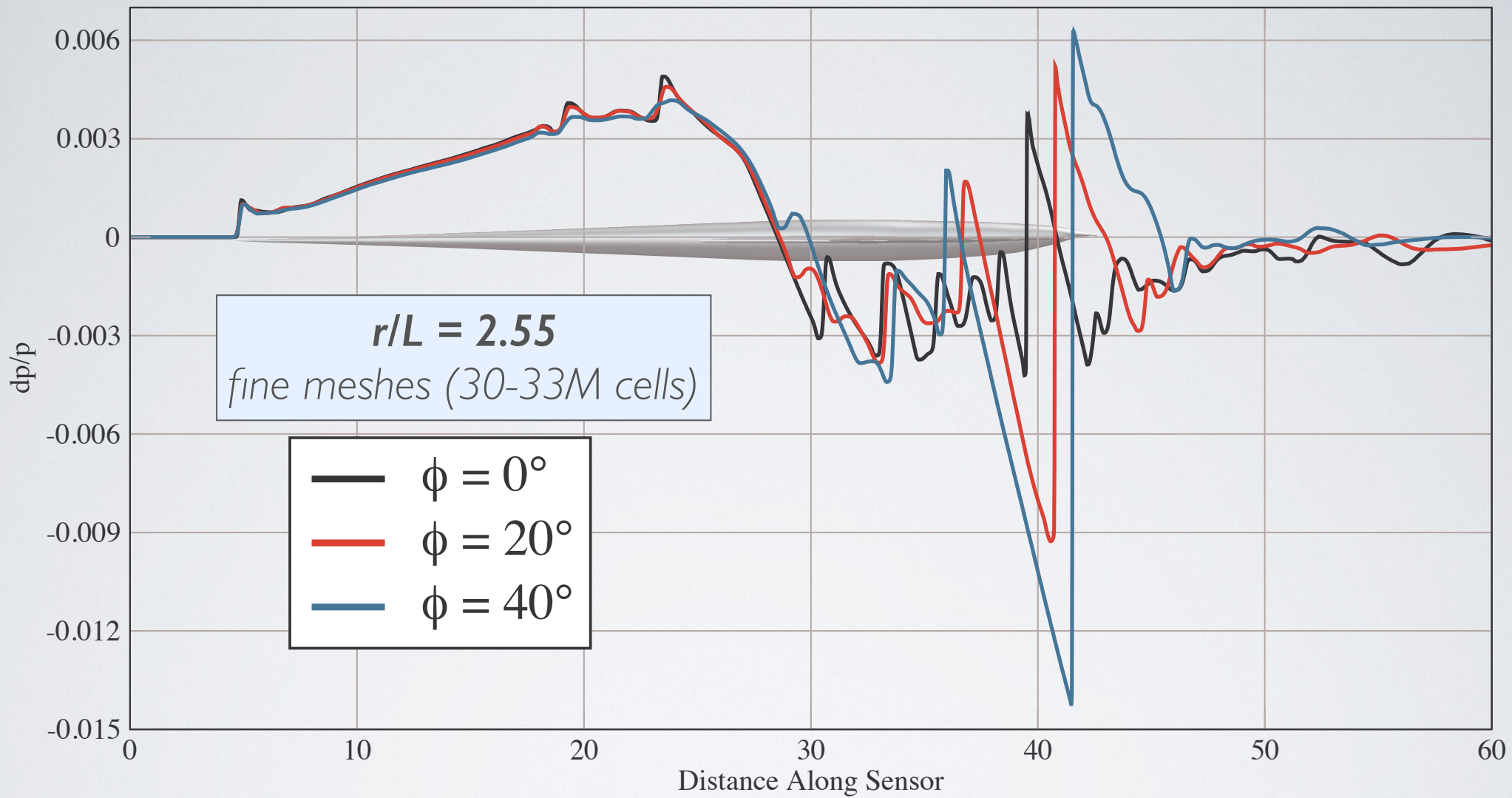
- ▶ 30-33M cells each
- ▶ Solved in parallel on six 28-core Broadwell nodes
- ▶ 2hr30min wall-clock time (*flow solution + all meshing, adjoint solutions, error estimation, etc.*)





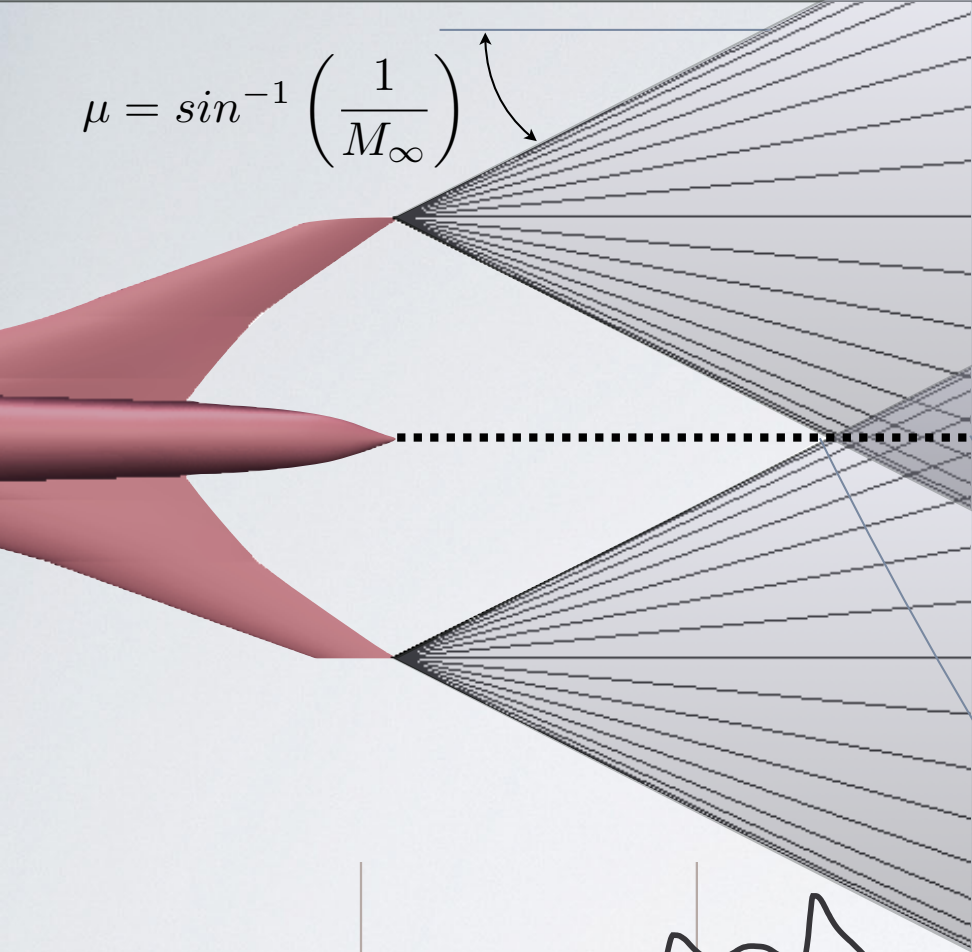






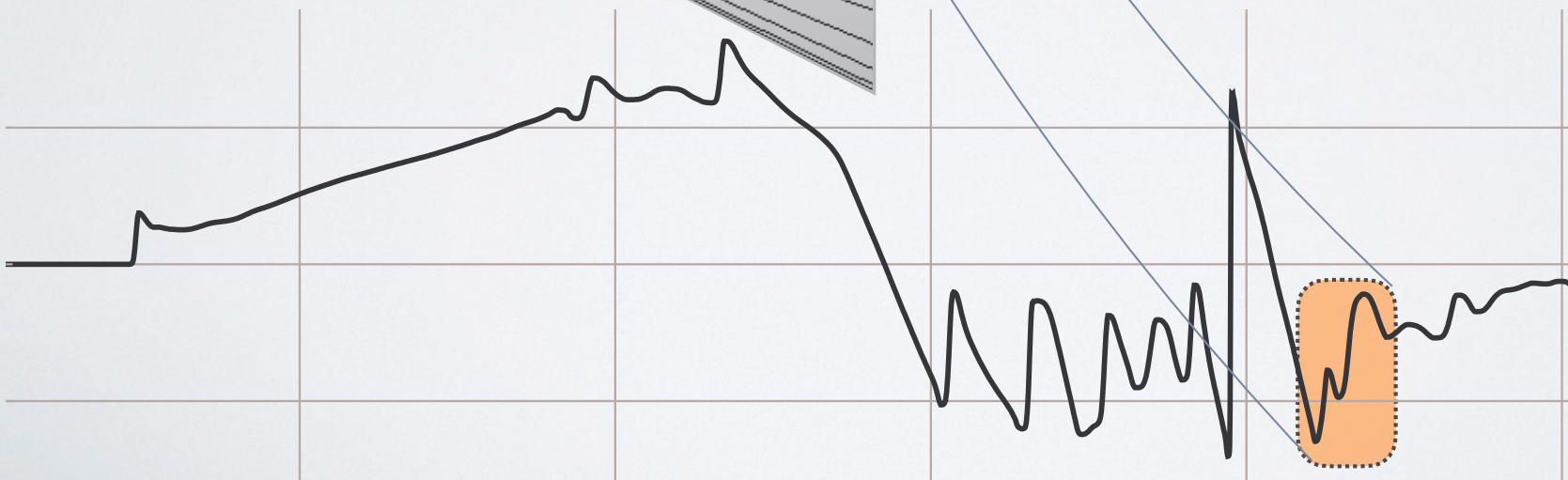
# JWB — ACCELERATING ADAPTATION

$$\mu = \sin^{-1} \left( \frac{1}{M_\infty} \right)$$



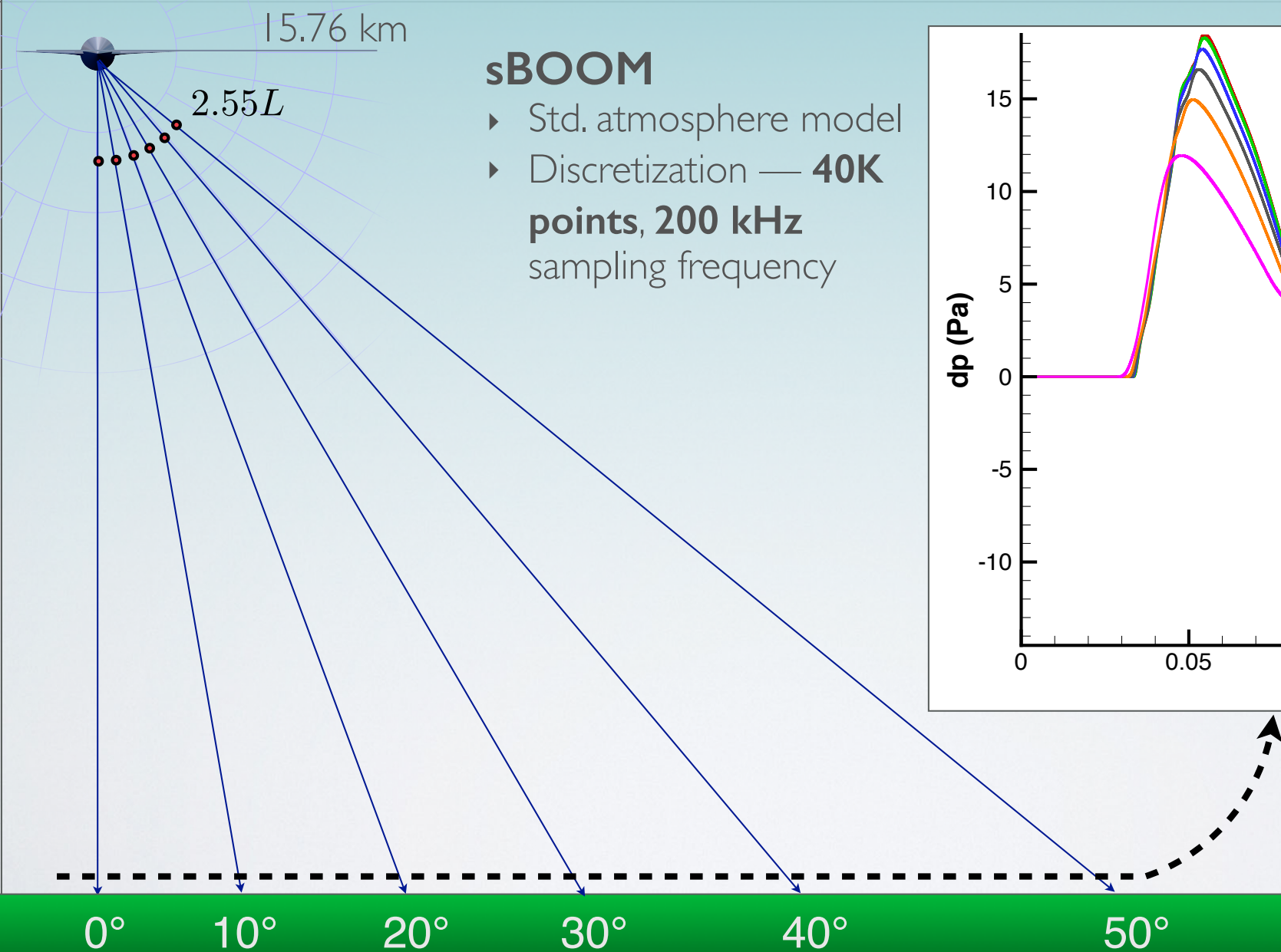
## Aft Fluctuations

- ▶ Precise behavior is very **mesh-sensitive**
- ▶ **Hypothesis: Wing tip vortices interacting with wake?**
  - ▶ Consistent with Mach cones
  - ▶ Localized adjoint solution traces back to wing-tips.



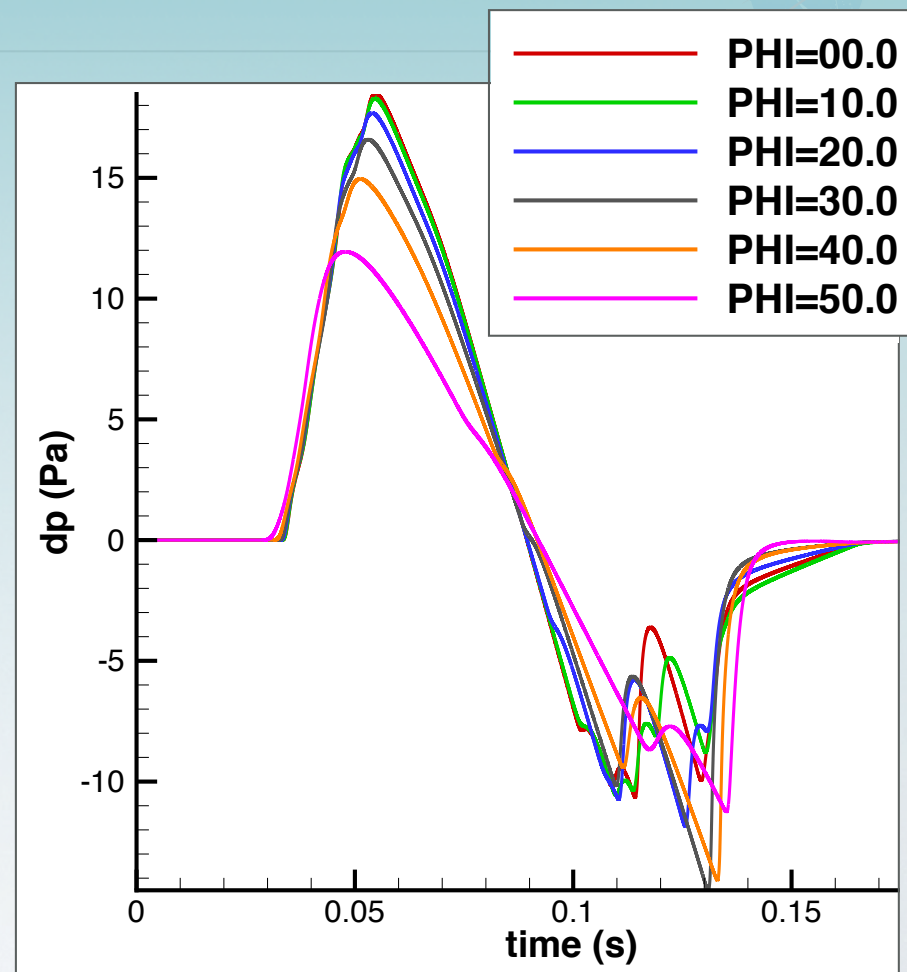


# JWB — PROPAGATION

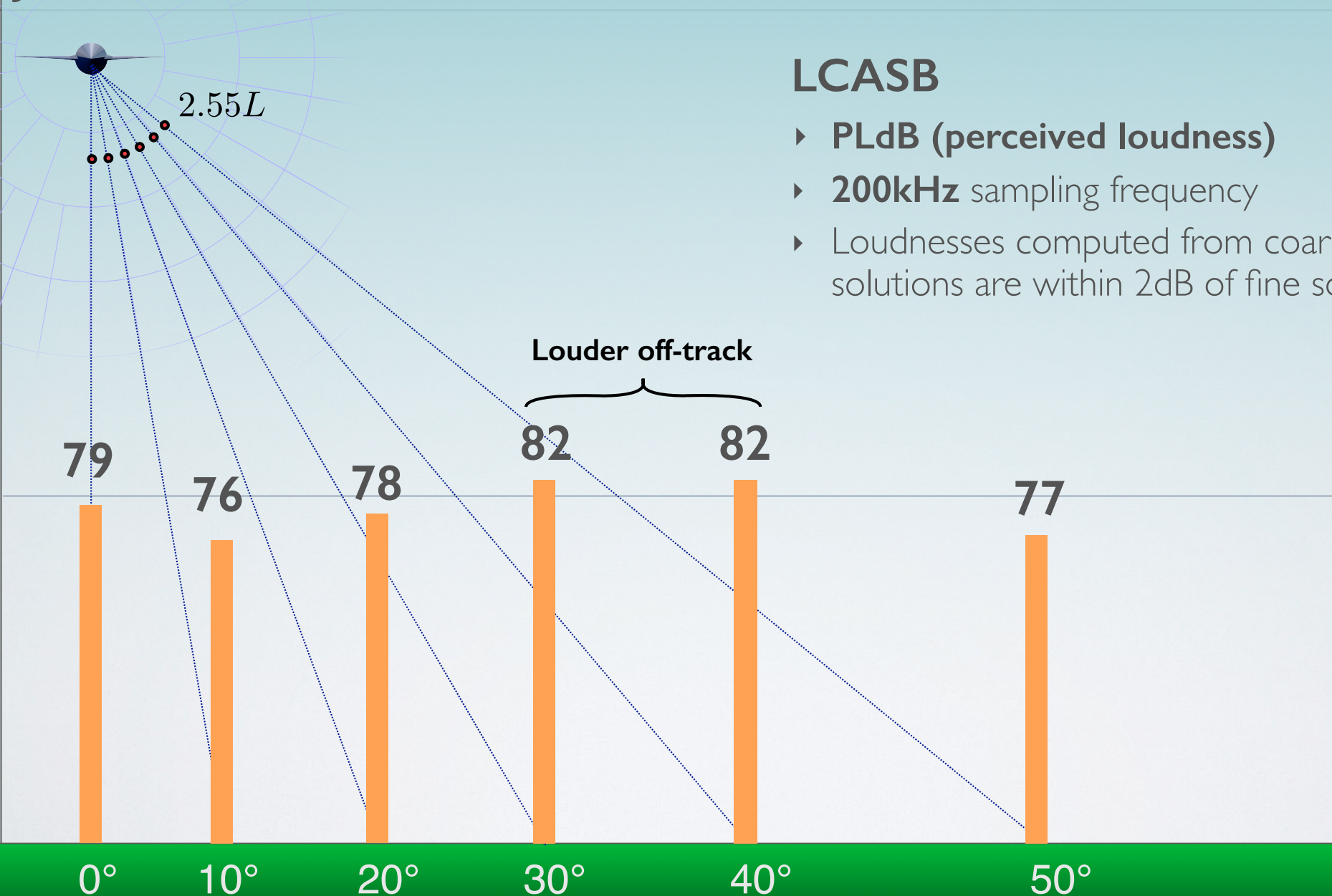


## sBOOM

- ▶ Std. atmosphere model
- ▶ Discretization — **40K points, 200 kHz** sampling frequency



# JWB — GROUND LOUDNESS

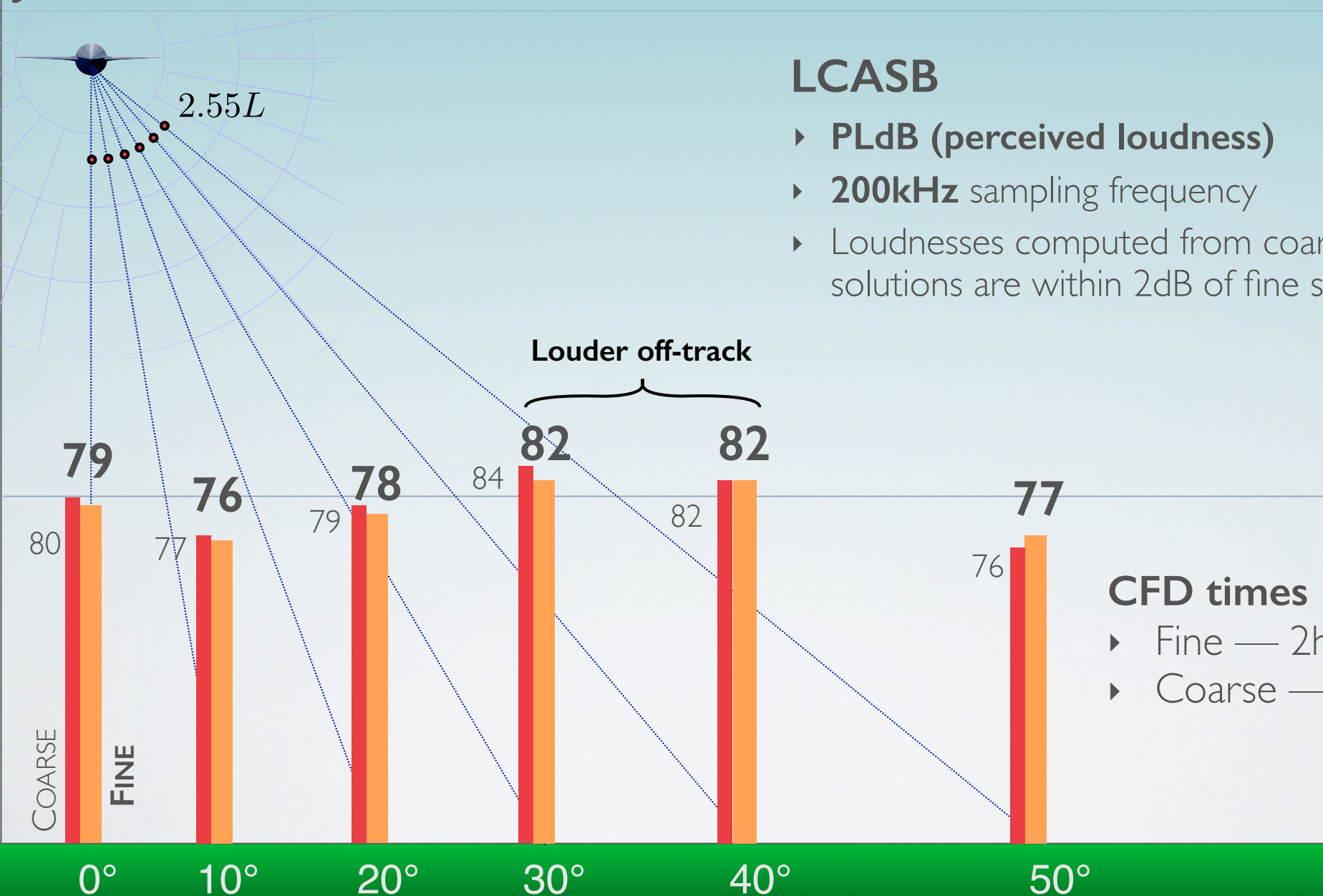


## LCASB

- ▶ **PLdB (perceived loudness)**
- ▶ **200kHz** sampling frequency
- ▶ Loudnesses computed from coarse solutions are within 2dB of fine solutions



# JWB — GROUND LOUDNESS



## LCASB

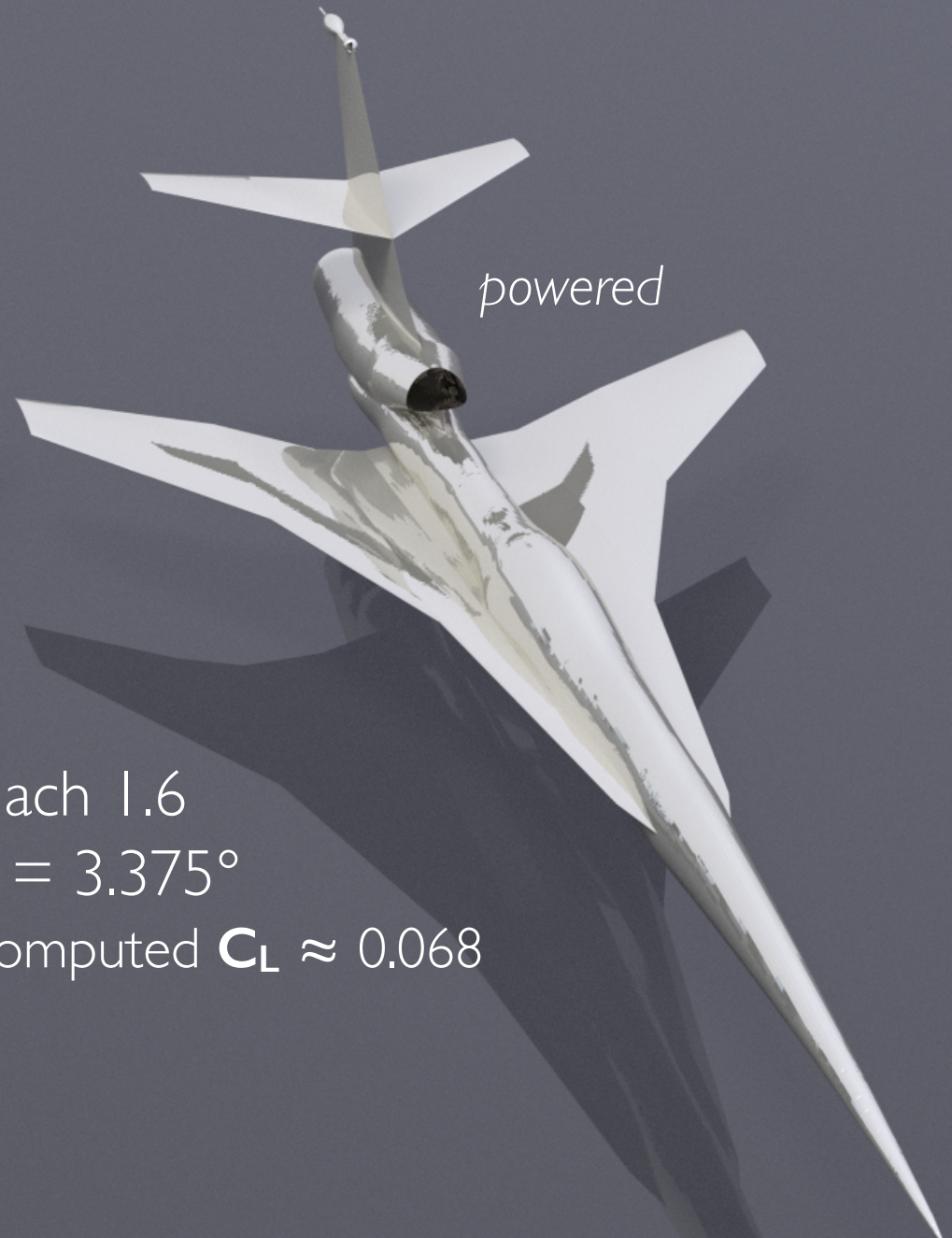
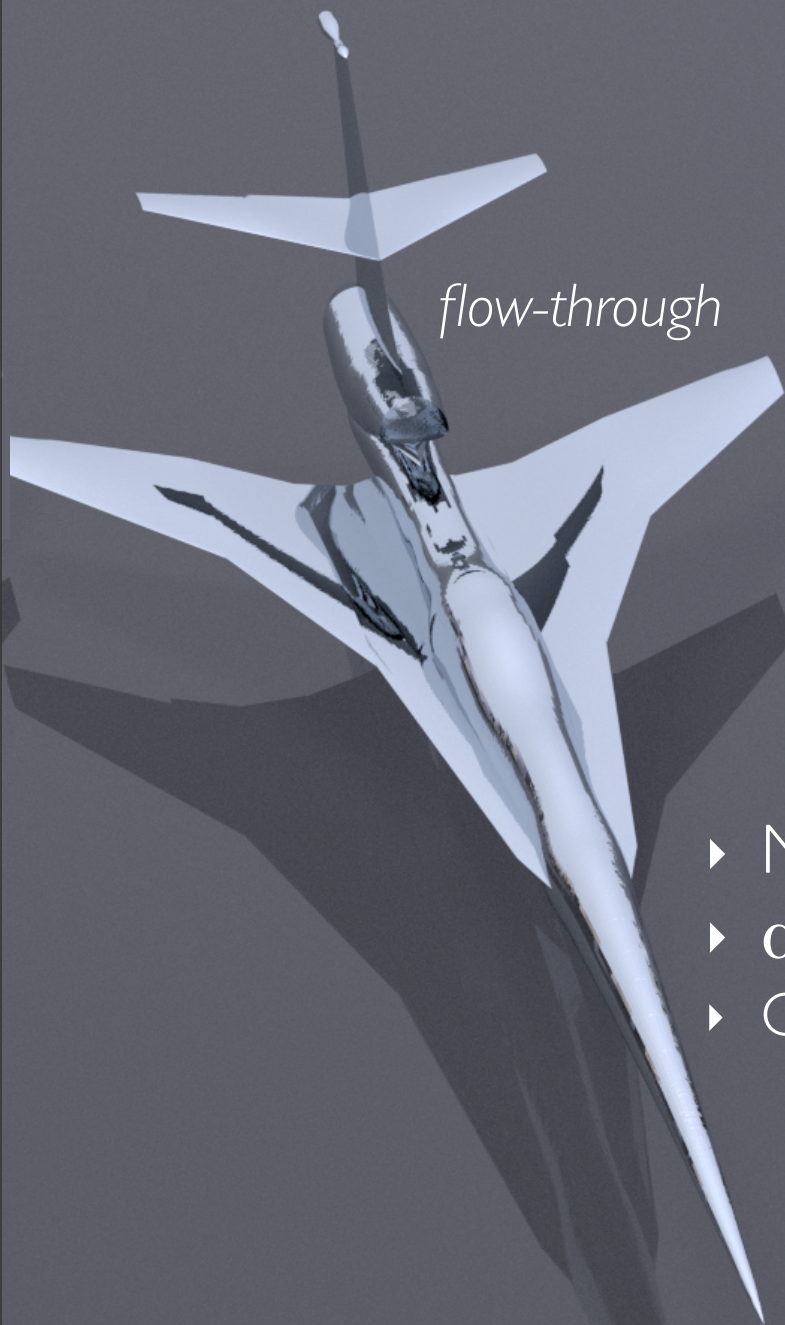
- ▶ **PLdB (perceived loudness)**
- ▶ **200kHz** sampling frequency
- ▶ Loudnesses computed from coarse solutions are within 2dB of fine solutions

## CFD times

- ▶ Fine — 2hr30min
- ▶ Coarse — 23min



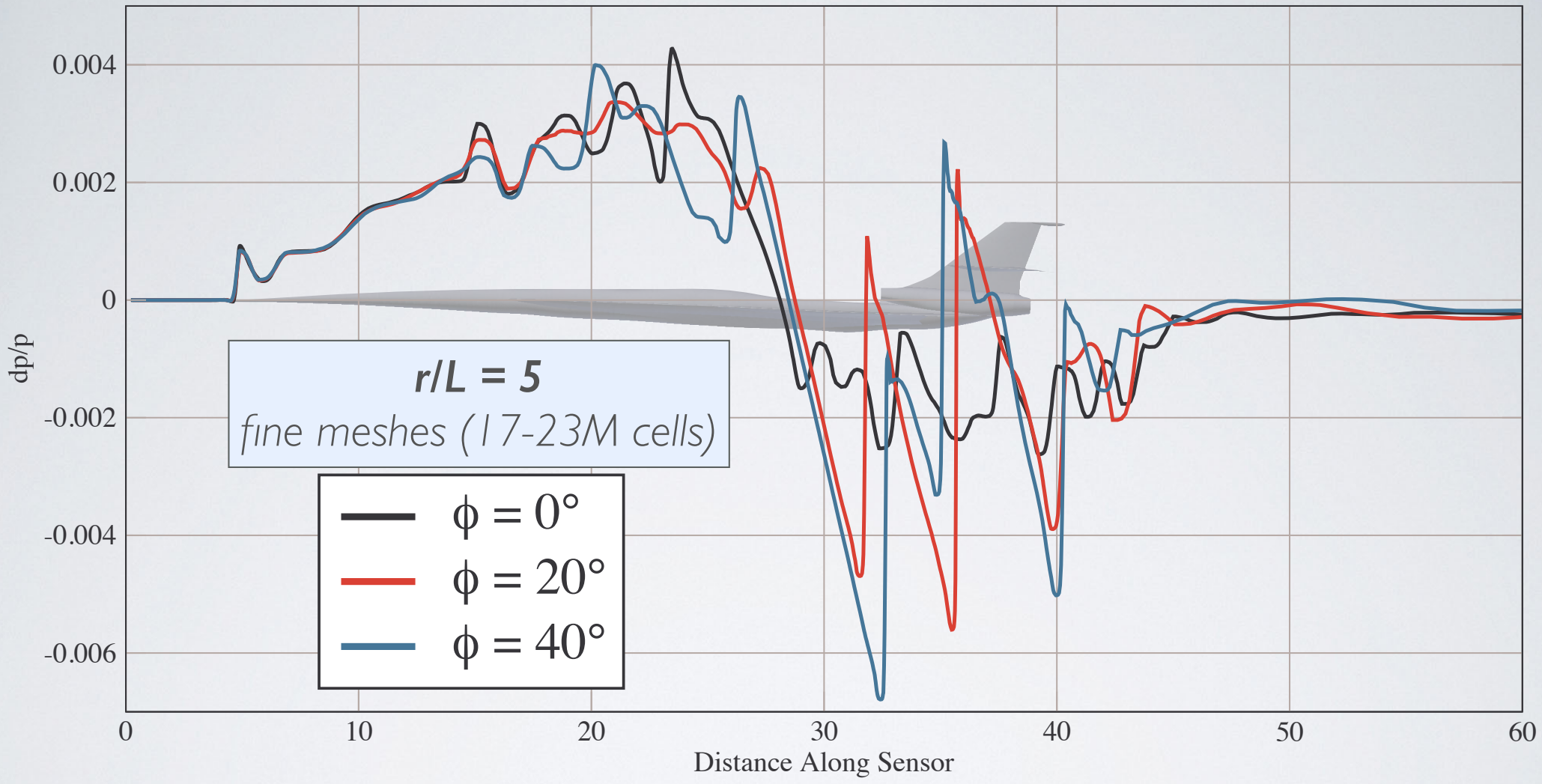
# C25-D



- ▶ Mach 1.6
- ▶  $\alpha = 3.375^\circ$
- ▶ Computed  $C_L \approx 0.068$



# C25-D FLOWTHRU — SIGNATURES

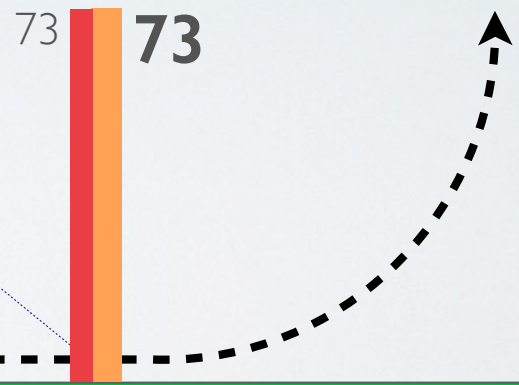
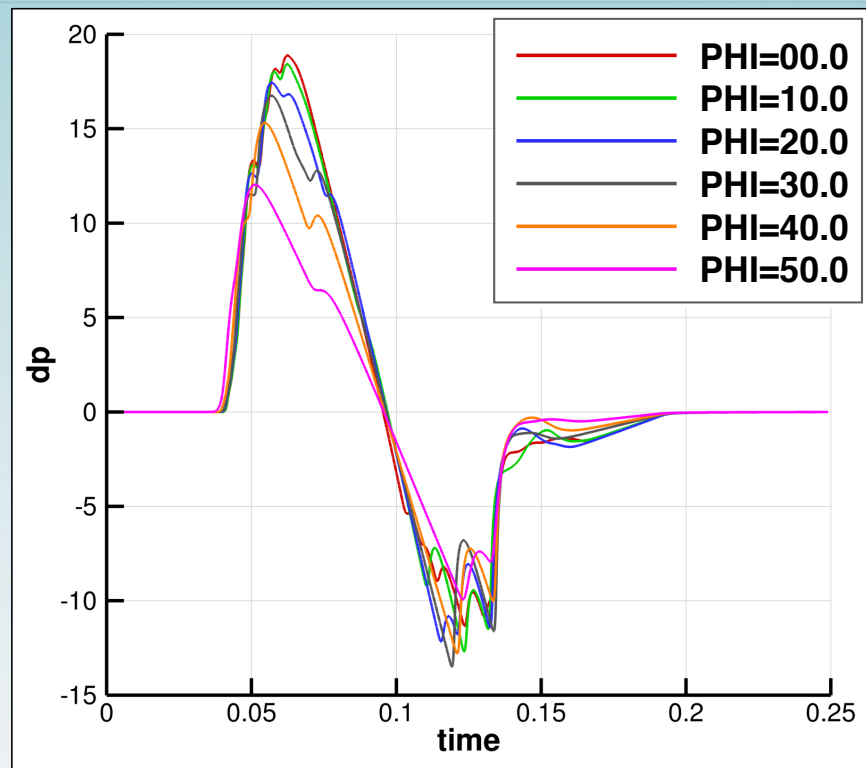
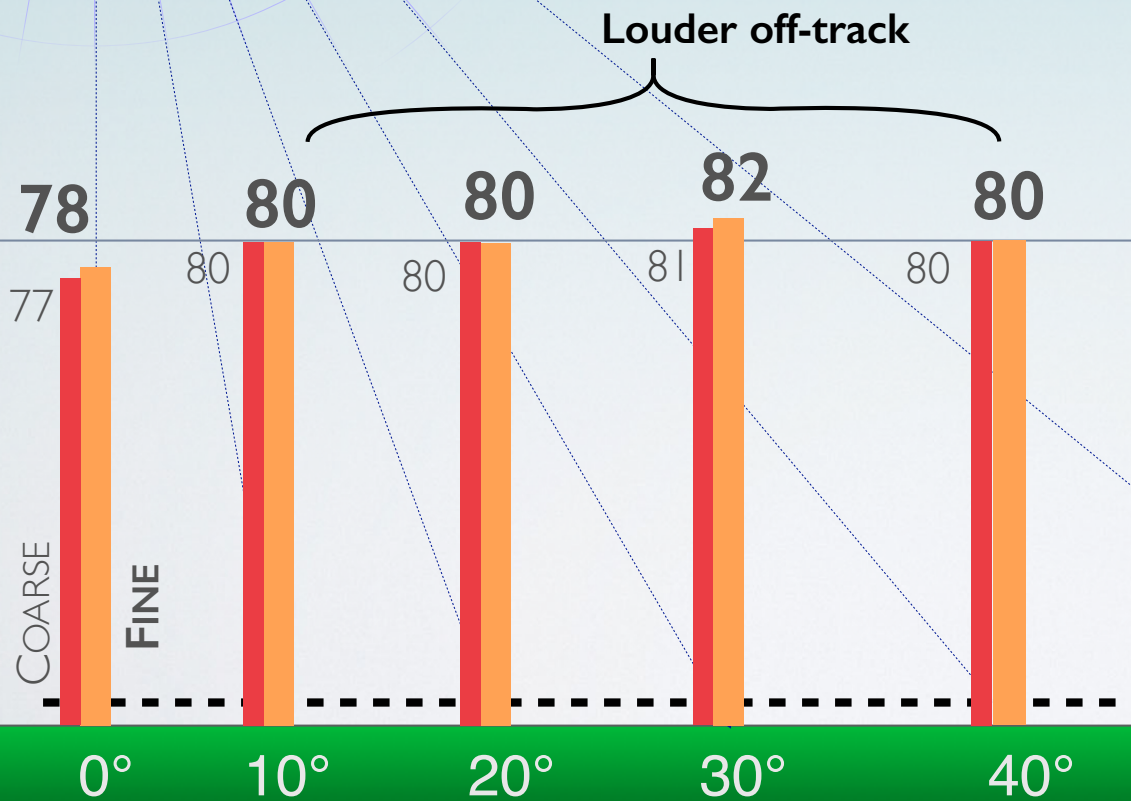


# C25-D FLOWTHRU — PROPAGATION



## sBOOM + LCASB

- ▶ Std. atmosphere model
- ▶ Loudnesses computed from coarse solutions are within 1 dB of fine solutions





# C25-D POWERED

Re-contoured fuselage  
and tail bulb

Inlet Conditions

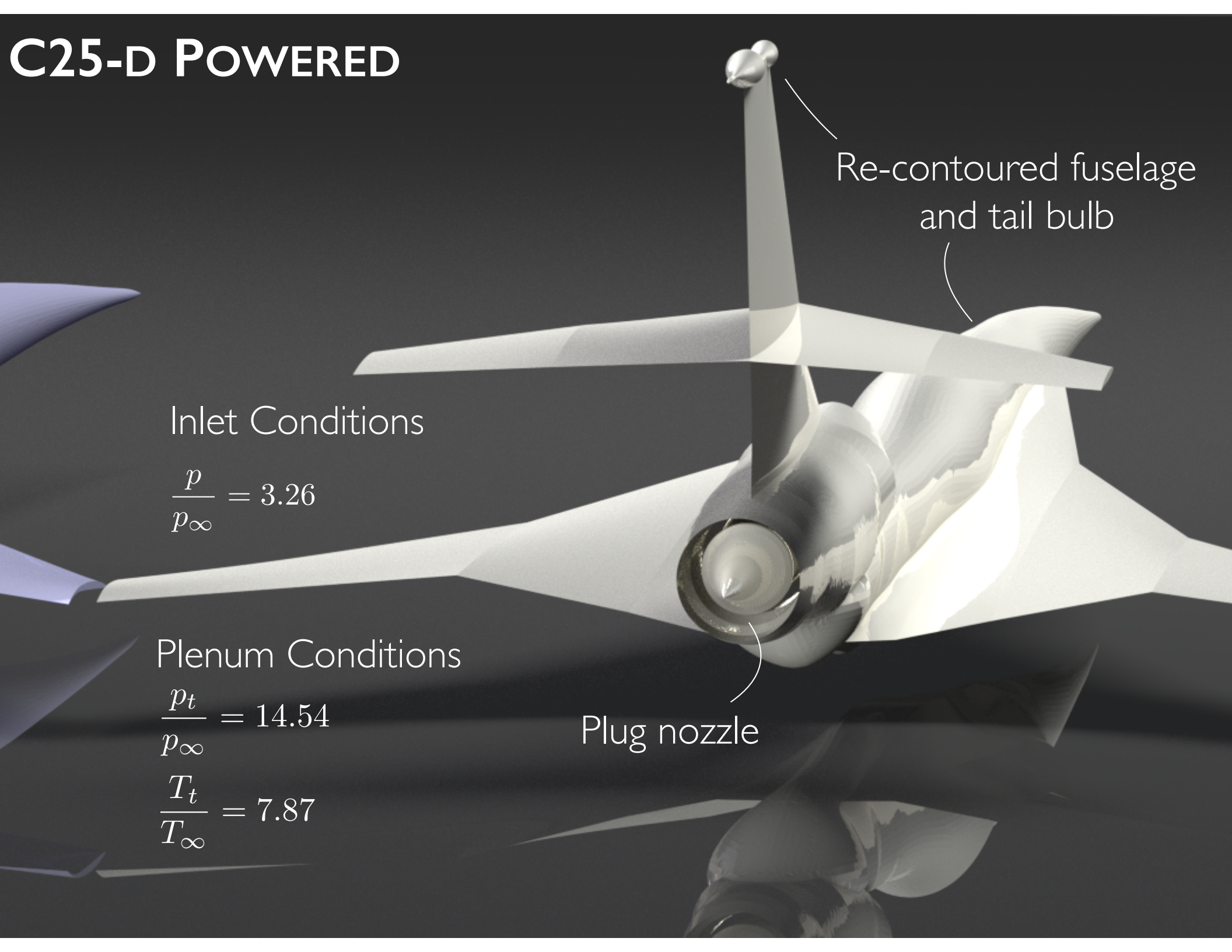
$$\frac{p}{p_{\infty}} = 3.26$$

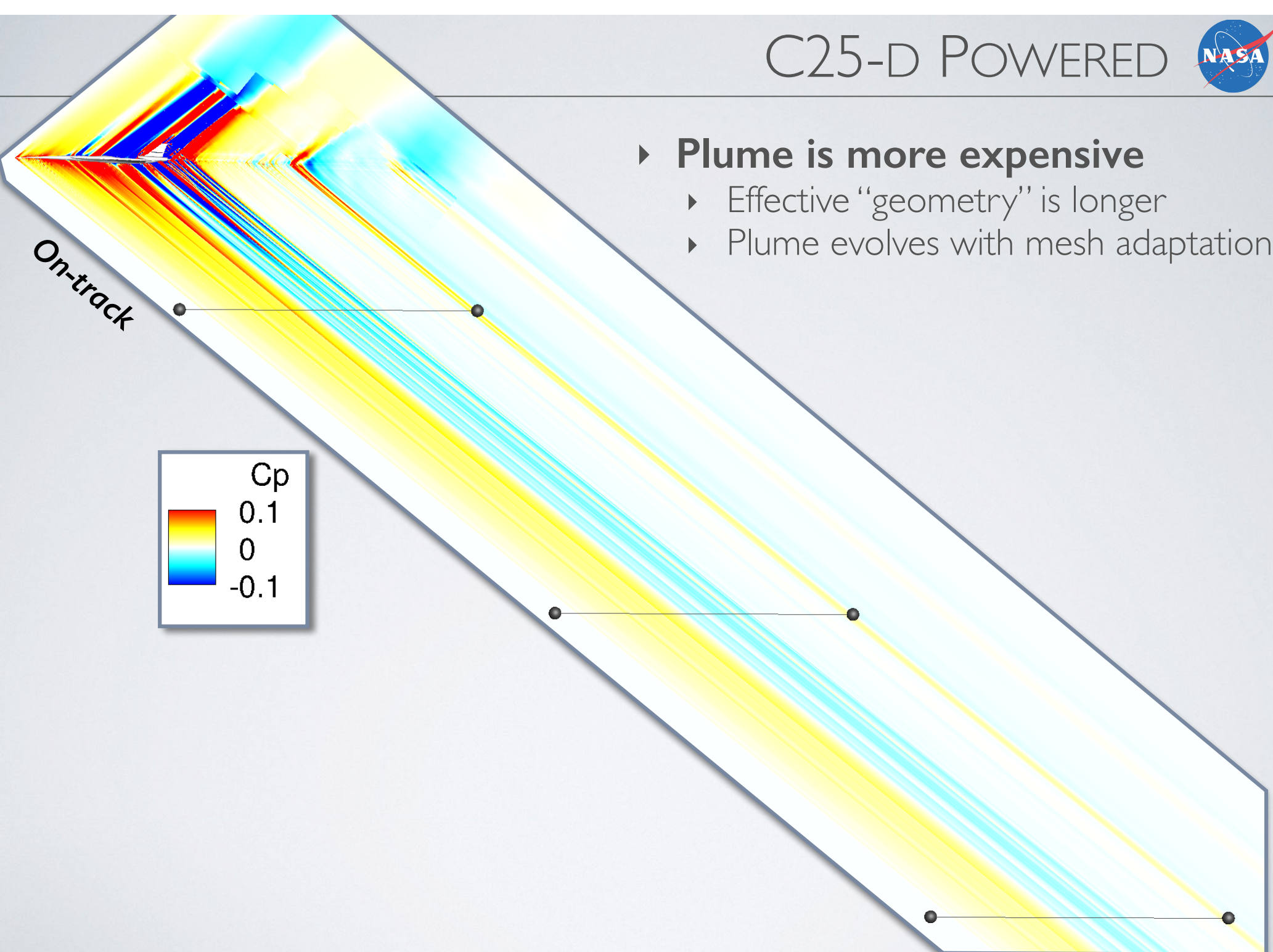
Plenum Conditions

$$\frac{p_t}{p_{\infty}} = 14.54$$

$$\frac{T_t}{T_{\infty}} = 7.87$$

Plug nozzle



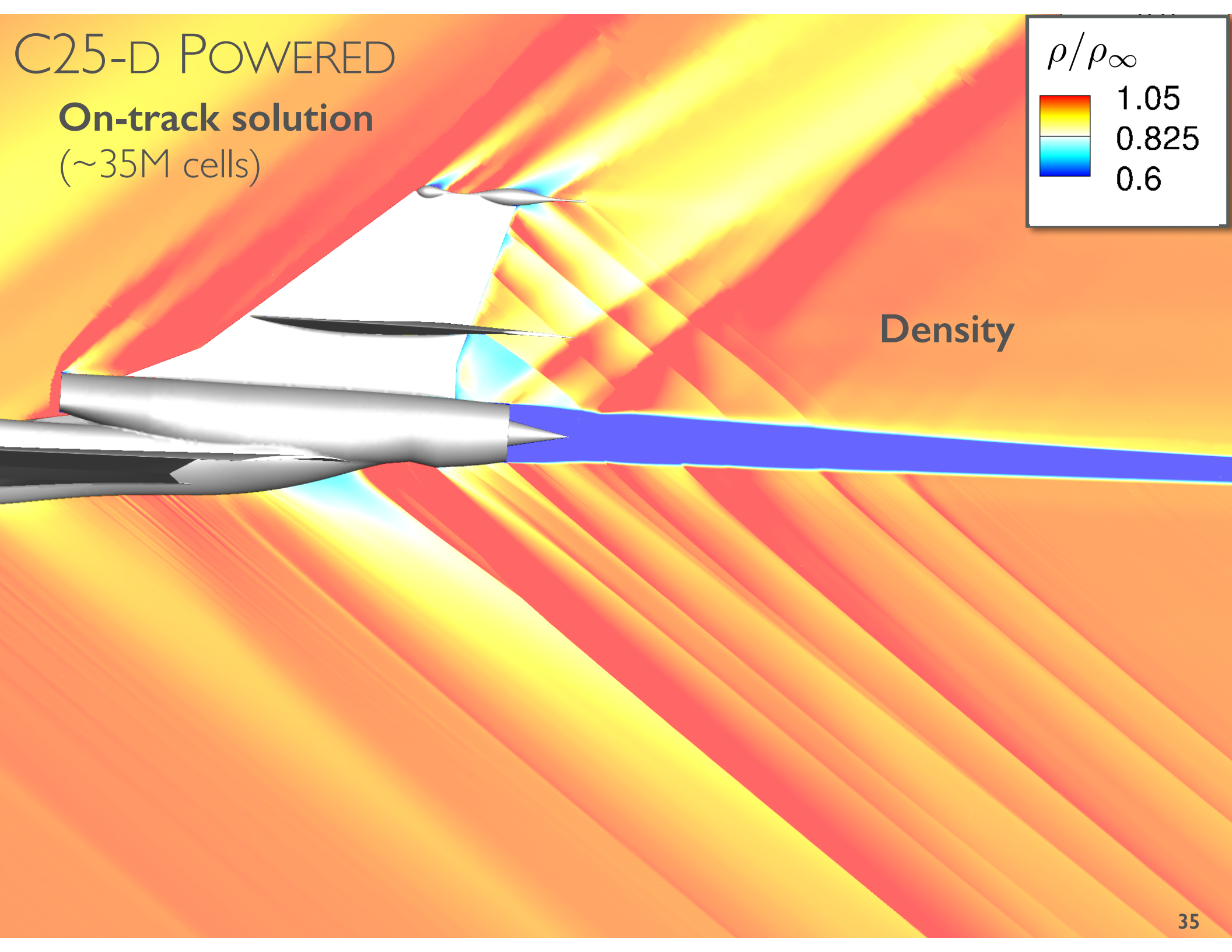




# C25-D POWERED

**On-track solution**

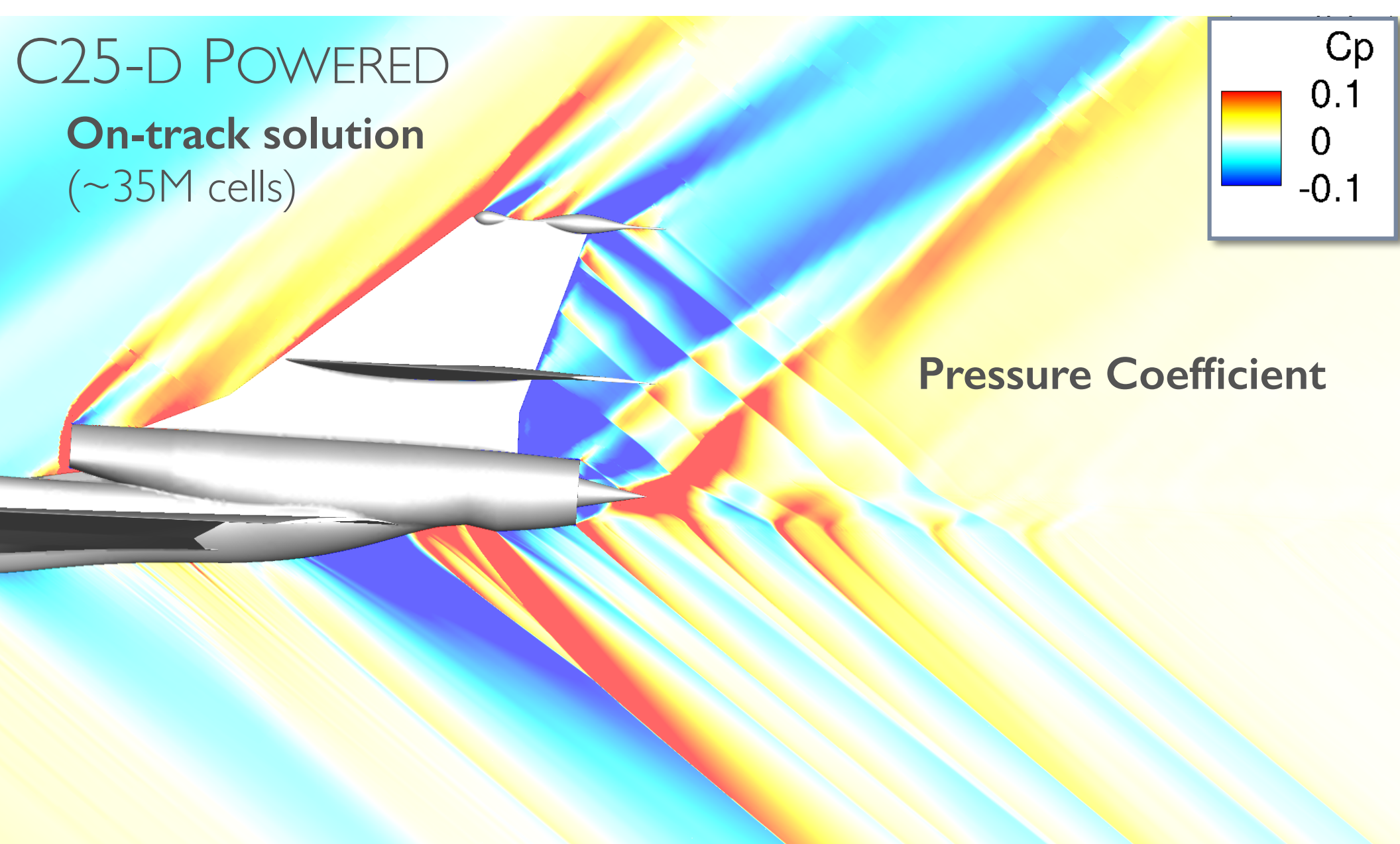
(~35M cells)



# C25-D POWERED

**On-track solution**

(~35M cells)

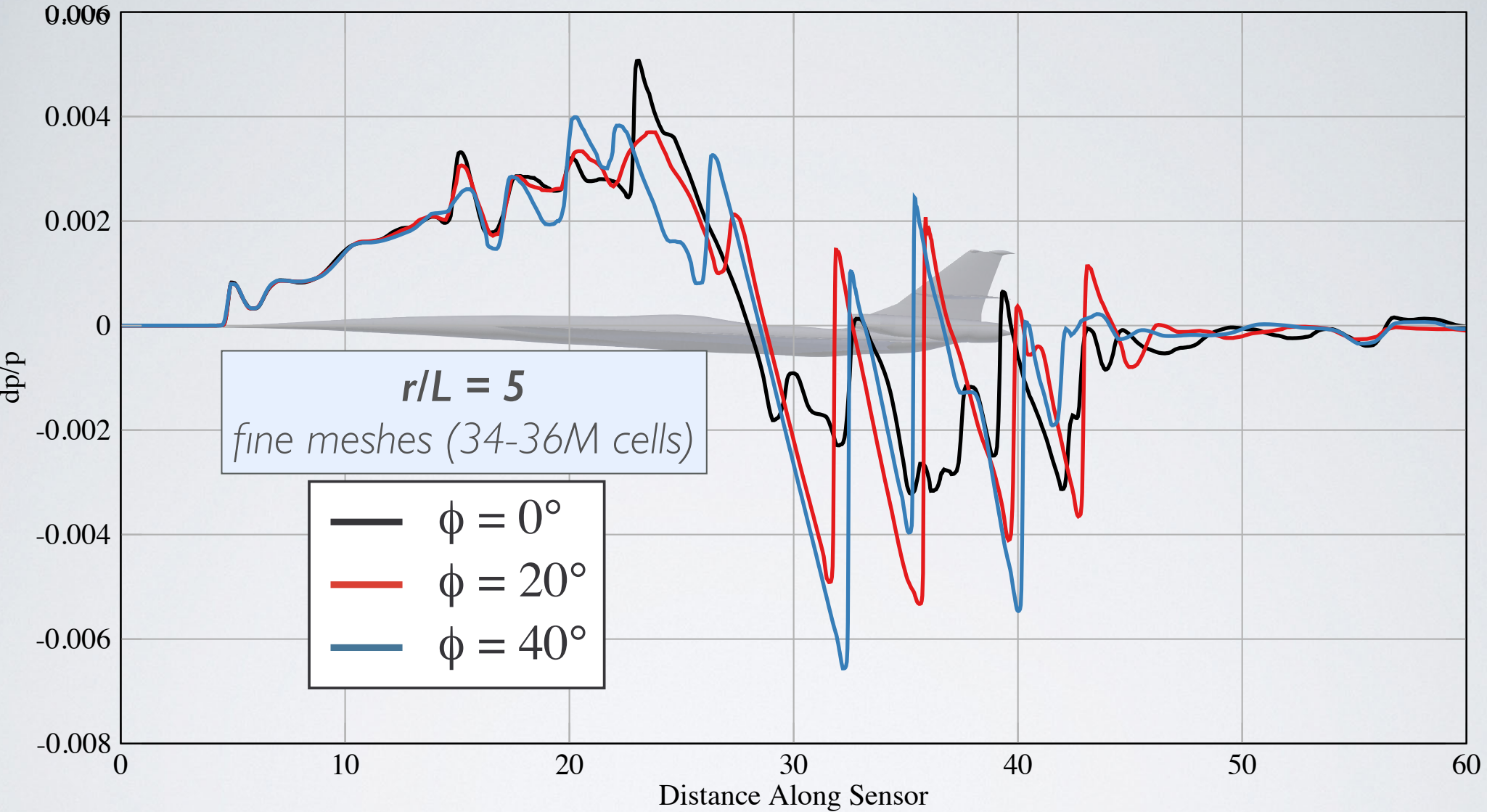
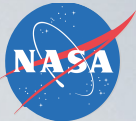


**Pressure Coefficient**

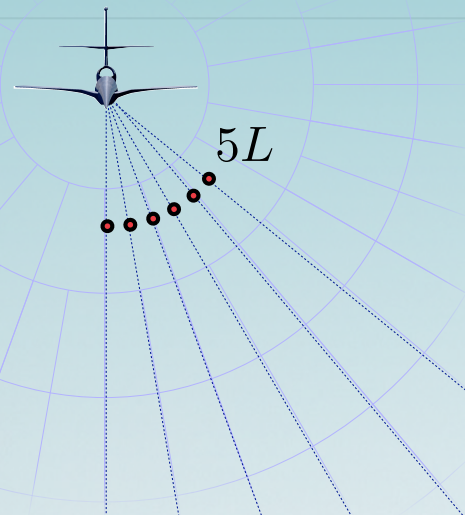
- ▶ **Each azimuth** — 35M cells — 4hr 30min on 28-core Broadwell node
  - ▶ *Includes flow solution + all meshing, adjoint solutions, error estimation, etc.*



# C25-D POWERED — SIGNATURES

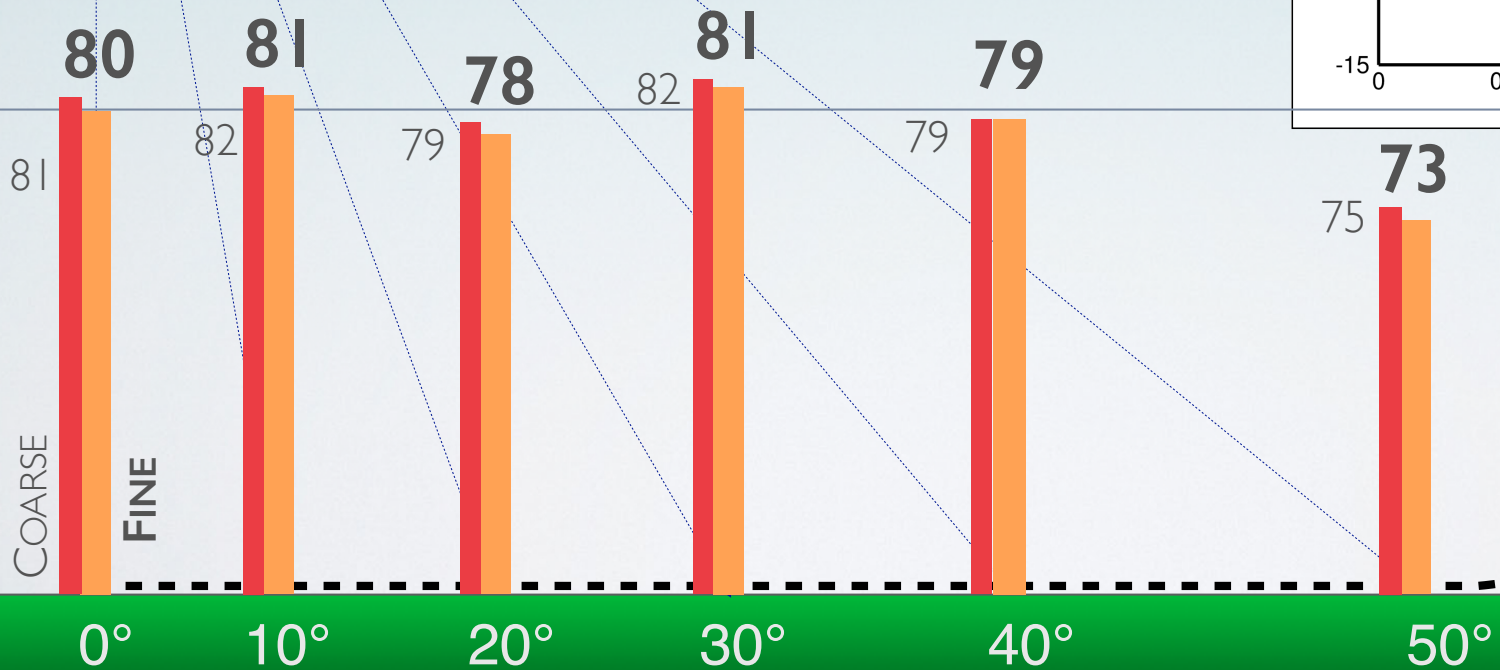
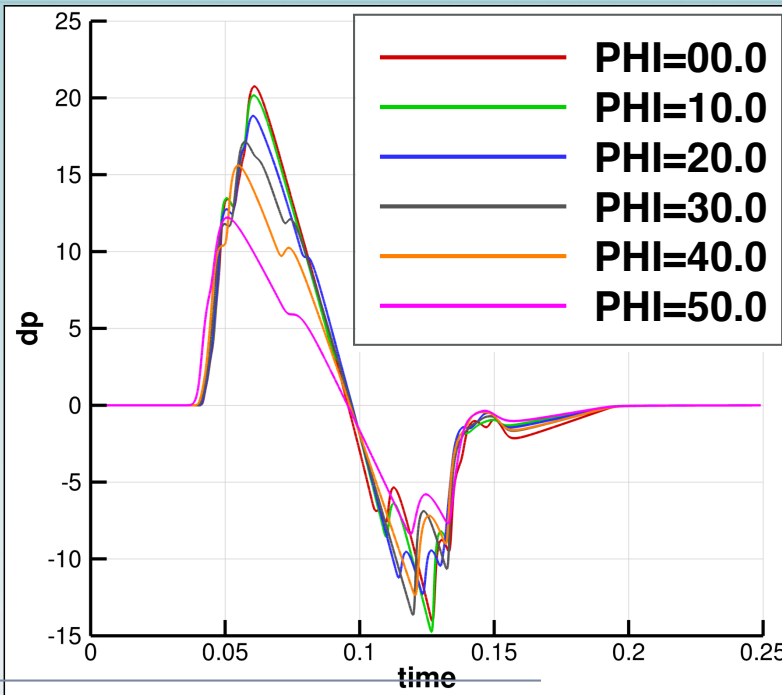


# C25-D POWERED — PROPAGATION



## sBOOM + LCASB

- ▶ Std. atmosphere model
- ▶ Loudnesses computed from coarse solutions are within 2dB of fine solutions





- ✓ **Solvers and Adaptive Meshing**  
*Axibody results as demo*
- ✓ **Multi-azimuth Problems with Cartesian Meshes**
- ✓ **Selected Workshop Results**
  - ▶ **Mesh Size and Cost**
  - **Conclusions**

- ▶ All results and timings were performed on **single Broadwell nodes**:
  - ▶ 28 cores, 120GB memory — roughly equivalent to a powerful Linux workstation
- ▶ For multi-azimuth cases, **azimuths computed in parallel on 6 separate nodes** on NASA's Pleiades supercomputer

Total costs for <b>all problems, all azimuths</b>	
<b>1450</b>	CPU-hours
<b>52</b>	Node-hours
<b>4hr 30min</b>	Max wall-clock time

### **Break-down:**

- 1x Axibody — (1hr 17min)
- 6x JWB — (2hr 28min each)
- 6x Flowthru — (1hr 30min each)
- 6x Powered — (4hr 27min each)

- ▶ **50-50 split** — Typically, about half the time is spent on adaptive meshing and half on the final flow solution.



## Finest Grid Comparison

Cell Counts	Cart3D Adapted		Workshop Inviscid
	per azimuth (avg)	Total*	
Axibody	26M	<b>26M</b>	<b>56M</b>
JWB	32M	<b>190M</b>	<b>18M</b>
C25-d Flowthru	19M	<b>114M</b>	<b>104M</b>
C25-d Powered	35M	<b>209M</b>	<b>52M</b>

\* Sum of cell counts across azimuths

# HIGHLIGHTS

- ▶ Spurious high-frequency oscillations from irregular surface geometry discretization.
- ▶ **Splitting azimuths greatly accelerates process:**
  - ▶ Run in parallel
  - ▶ Align mesh to azimuth → stretch to high AR
- ▶ JWB exhibits mesh-sensitive wake behavior — requires lots of resolution.
- ▶ Perceived loudness from coarse and fine CFD solutions differ by less than 2 PLdB for all cases and azimuths.

